



TERMINAL AREA DELAY AND FUEL CONSUMPTION ANALYSIS

ARTHUR G. HALVERSON GORDON JOLITZ

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JANUARY 1979 FINAL REPORT



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### PREFACE

The authors wish to express sincere appreciation to Messrs. Richard Soper and Thomas Choyce of the Analysis Branch, ANA-220, for their diligent efforts during the conduct of this study. In particular, the software system developed by Soper and Choyce will have ever increasing application in the analysis of field-derived data.



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#### EXECUTIVE SUMMARY

The purpose of this project was to develop estimates of excess mileage flown in the terminal area, to estimate excess fuel burn due to air traffic control (ATC) delay maneuvers, and to develop a method to analyze the effect of future ATC concepts to reduce delay. The results discussed in this report are from an analysis of ARTS track data collected during the 1974-1975 time period as a data base for the ATC/Airborne Collision Avoidance System (ACAS) Compatibility Analysis project. The data base consists of 48 hours of Advanced Radar Terminal System (ARTS) tracks; 12 hours each from Chicago, Miami, Los Angeles, and Washington. These data were collected in accordance with the criteria established for the ATC/ACAS analysis which were not completely in consonance with criteria required for delay analysis. Regardless of this limitation, the study yielded several significant findings, such as:

- 1. It was found that ARTS track data provides a viable medium to derive credible estimates of terminal area delay and excess fuel consumption. However, raw data as recorded by the ARTS computers are characterized by anomalies, spurious data, and other vagaries. Therefore, the process used to derive these estimates should provide for manual intervention at appropriate points; otherwise, the results may be misleading. The methodology applied in this project permitted manual review, evaluation, and editing of ARTS track data. Human judgment was applied in areas where decisions through program logic would be suspect. The methodology proved to be effective and economical and should have broad application in future analysis of ARTS data.
- 2. In addition to excess fuel consumption that results from holding, pathstretching, and speed control delay, other sources of excess fuel consumption
  were revealed from analysis of ARTS track data. These include excess route
  mileage due to local procedures and increased flight time and fuel flow due
  to premature descent from cruise altitude. In connection with the latter, the
  data were collected before profile descent procedures were implemented
  (reference 1). This, obviously, should have an effect on that source of
  excess fuel consumption.
- 3. It was found that when delay is required, the proper application of speed control and early descent is a fuel-efficient method of absorbing the delay. However, speeds requiring the use of flaps should be avoided, if possible, and early descent from cruise altitude should not be a matter of routine practice, but, rather, should be used only when delay is required.
- 4. Rigid procedures, where an attempt is made to absorb all delay by highaltitude holding, were not supported by this analysis. However, efforts to develop fuel-efficient scenarios to absorb delay are encouraged. As inputs to these efforts, the following strategies are provided, in descending order of priority, depending upon the amount of delay required, the predictability of the ATC system, navigational accuracy, and other factors:
- a. Reduced speed while descending from enroute altitude to metering fix altitude,

- b. Reduced speed at enroute cruising altitude,
- c. Descent to an appropriate lower cruising altitude to effect further speed reduction (this requires further study),
- d. Reduced speed between the metering fix and the approach gate, but not below clean-flap configuration,
  - e. High-altitude holding,
  - f. Path-stretching vectors, and,
  - g. Lower speeds near the approach gate if needed for fine-grain control.

It should be noted that the above strategies are not in complete consonance with the scenarios depicted in reference 2.

5. The need for fuel-efficient delay-absorbing strategies is clearly evident from the delay and excess fuel consumption data derived in this project for the Chicago O'Hare (ORD) airport. The average delay for the 635 ORD arrival tracks analyzed was computed to be approximately 10 minutes. The excess fuel consumption due to this delay was estimated to be 1,055 pounds (1b) (157 gallons) per track. Assuming that the current traffic levels at ORD are at least equal to those in the data base, an annual estimate of delay and excess fuel consumption for arrival aircraft at ORD can reasonably be placed in the area of 2.5 million minutes and 40 million gallons, respectively. At average 1978 prices, the cost of this delay to the users of the ATC system is estimated to be in the range of 33 to 40 million dollars.

Although the data from Miami (MIA), Los Angeles (LAX), and Washington (DCA) produced some enlightening results, no attempt was made to extrapolate annual estimates from these data. It was felt that the data samples from these airports were not sufficiently representative of the periods during which delay normally occurs. Also, annual estimates at these airports would require a thorough analysis of traffic loads, weather, and other factors. At ORD, on the other hand, traffic generally remains at high levels from 9 a.m. to 9 p.m., daily. In addition, the ORD data samples used in this analysis were collected during periods of instrument flight rules (IFR) conditions as well as visual flight rules (VFR) conditions and, also, the principal runway configurations were represented in the data base.

6. The criteria to be followed when collecting data for delay and excess fuel analyses are critical. The samples should include a representative range of weather conditions and runway configurations and each sample should be of sufficient duration to capture the oscillating effect of traffic demand (peaks and valleys) on delay and excess fuel consumption. Also, terminal area delay may start to accrue while aircraft are still well into the enroute area. Therefore, for a more complete analysis of terminal area delay, data should also be collected from selected sectors of the air route traffic control center (ARTCC).

#### INTRODUCTION

This project was established to support the Federal Aviation Administration (FAA) Advanced System Engineering Program (ASE) which was formulated by the Office of Systems Engineering Management (OSEM). Specifically, the objectives of this work were to:

- 1. Develop a data base of actual miles flown in the terminal area versus nominal and minimum route lengths which can serve as a measure of efficiency for future ATC system design concepts;
- 2. Derive first-order estimates of excess fuel consumed in the terminal area due to holding, path-stretching vectors, ATC procedures, and other factors associated with delay; and
- 3. Develop a methodology for the analysis of terminal area delay and excess fuel consumption which can support advanced concept development efforts on an as-needed basis.

Initially, the work was to include the enroute area and, where possible, actual runway-to-runway flight mileages were to be compared with fuel-conservative, direct routes as defined in the ASE Program Plan. However, due to the magnitude of the requisite data collection and reduction efforts, the scope of the effort was subsequently reduced to include only representative terminal areas.

It was the general consensus that the best source for analysis of terminal area mileage and delay was track data being recorded online by ARTS III. It was recognized, however, that effective analysis of the ARTS track data requires a substantial amount of computer software that was not available at the time. As an initial step, therefore, it was decided to use an available data base of ARTS tracks as a vehicle to develop the required software and other methodology. This data base was developed during an ATC/ACAS Compatibility Analysis (reference 3) and consists of ARTS data from Chicago, Miami, Los Angeles, and Washington.

This report discusses estimates of delay and excess fuel consumption derived during this initial phase. In addition, a general description of the methodology developed to conduct these analysis is provided. Detailed descriptions of the computer software is contained in separate documentation.

### METHOD OF APPROACH

#### DATA BASE.

With the advent of ARTS and an associated data recording capability, it became possible to perform comprehensive analysis of real world operations in the terminal areas. It is recognized, however, that a data collection program designed to meet specific analytical objectives is a costly and time-consuming task. Moreover, for a variety of reasons, data recorded by ARTS require a considerable amount of processing and editing before being effectively applied to operational analyses. In the interest of economy and expediency, therefore, it was decided to use data for this project that had been collected earlier for the purpose of investigating the interaction between the ATC system and a proposed ACAS (see reference 3).

The data base developed during that study, referred to as the "Field-Derived Data Base," or FDDB, had many of the features needed to meet the objectives of this study. As shown in table 1, the FDDB consists of 48 1-hour data samples--12 each at Chicago, Miami, Los Angeles, and Washington.

In addition to being readily available, this data base had undergone several levels of processing, most of which is required for any type of analysis of ARTS data. Track data had been smoothed between beacon acquisition points, missing altitude data had been added, and most anomolies had been removed. There were, however, some disadvantages in using the FDDB for this study. The data were collected during the 1974/1975 time period and therefore may not completely reflect current procedures and traffic loads. Also, the data samples were selected to meet the analytical objectives of the ATC/ACAS study which were not completely consonant with delay and fuel consumption analysis. For example, more periods when delay is expected to occur, such as prolonged IFR conditions, would have been desirable. For delay analysis, samples of longer than 1-hour duration are needed to capture the effects of oscillating traffic demand on the terminal. Also, delay due to terminal conditions may start to accrue while aircraft are still under center control. Therefore, for more complete analysis of terminal area delay, it is necessary to also collect data from the enroute sectors which feed traffic into the terminal area.

Regardless of these and other limitations, it was felt that the FDDB provided a good starting point for investigating delay and excess fuel consumption that result from many interrelated factors in terminal air traffic control. In particular, the FDDB offered an ideal instrument for the development of computer programs and other methodology needed to conduct a credible analysis of these data. Further, although many problems had already been eliminated, analysis of the FDDB revealed that reduction of delay data from ARTS tracks without provision for manual intervention could yield highly questionable results. Accordingly, the process developed consists of a series of sequential steps, where each step provides the capability for manual interface with computer processing of the track data (see "Description of Methodology").

TABLE 1. CONTENTS OF THE FIELD-DERIVED DATA BASE (FDDB)

LOCATION	NO. OF 1-HOUR SAMPLES	AVG. HOURLY ARRIVAL RATE	AVG. HOURLY DEPARTURE RATE	AVG. HOURLY OPERATIONS RATE	WEATHER	ARRIVAL RUNWAY(S)
Chicago O'Hare	3	66	Note 1	Note 1	IFR	14L/14R
(ORD)	3	68	Note 1	Note 1	VFR	14L/14R
	6	66	Note 1	Note 1	VFR	27R/32L Note 2
Washington Mational	6	30	26	56	VFR	36
(DCA)	6	29	26	56	VFR	18
Los Angeles Intl. (LAX)	3	33	40	73	IFR	24/25
Note 3	3	36	38	34	VFR	24/25
	3	33	38	71	VFR	6/7
Miami Intl.	6	31	28	. 59	VFR	9L/9R
(MIA)	3	32	22	54	VFR	27L/27R
	3/	29	23	52	Note 4	Note 4

#### Note:

- Departure load was not a factor at ORD, due to use of independent runways.
- In one sample, aircraft were using 27R/27L for the first half of the hour, then the 27L traffic was changed to 32L.
- 3 An additional three 1-hour set was collected at LAX for a particular ATC/ACAS probe. Because the operations rates were typically low, they were not included in the analysis.
- 4 These samples were taken when thunderstorms were reported in the terminal area. Direction of landing was changed twice in one sample.

### DELAY MEASURES.

Before discussing the results of this study, it is important to clarify what is meant, herein, by "delay" and by "excess fuel consumption." Webster's New World Dictionary states that "delay implies the interference of something that causes a detainment or postponement." As applied to arrival aircraft (departures were not analyzed), the dictionary interpretation of "delay" would infer that delay is the result of one or more constraints imposed on the aircraft's movement that causes the landing time to be later than it would otherwise have been.

There are, of course, many constraints which cause arrival aircraft to encounter delay, several of which are associated with the operation of the aircraft itself. For example, cabin pressure management may preclude optimum descent gradient, and passenger comfort may limit aircraft maneuvering which could increase flying time. Other constraints causing delay can be the result of weather, terrain, noise abatement procedures, vortex phenomenon, and the like. Of primary concern in this study, however, are the constraints imposed by the ATC system during the normal performance of its mission—i.e., safe and expeditious movement of traffic. It should be pointed out that it is not the intent in this work to judge the performance of the ATC system. The sole purpose, on the other hand, is to provide objective data on how much delay accrues under differing circumstances and, further, to produce estimates as to how much excess fuel is consumed as a result of such delay. In this context then, "excess fuel consumption" is simply the fuel required over and above that which would have been consumed had the delaying constraint not been imposed.

In the normal course of air traffic control, there are three basic methods to effect delay: (a) holding in a racetrack pattern at a navigational fix, (b) path-stretching by radar vectors, and (c) speed control. Normally these delaying measures are applied in various combinations, depending upon local procedures, traffic demand, and many other factors. However, since each method impacts fuel consumption differently, it was decided to partition delay into components associated with each method. With this approach, the results can be more effectively applied in the development of fuel-conservative delay strategies.

DELAY DUE TO PATH-STRETCHING VECTORS. In a terminal radar environment, arrival aircraft normally do not navigate over a prescribed (i.e., charted) route from terminal entry point to touchdown. From feeder fix to the final approach course, navigation is primarily affected through radar vectors (headings) issued by air traffic controllers. From entry point to the feeder fix, navigation may be along a charted airway or very high frequency ommirange/tactical air navigation (VORTAC) radial, or may also be accomplished by vectoring. Moreover, radar vectors perform a dual function, i.e., navigation and separation. When separation requirements result in an increase in flying distance, the corresponding action is referred to as "path-stretching." Obviously this results in added flying time (i.e., delay) and an increase in fuel consumption; therefore, realistic measurement of path-stretching mileage was an essential requirement for this study. In order to derive this path-stretching distance, it was necessary to establish baseline routing by which to compare the actual

track distances from the FDDB. These routings are referred to as "nominal routes" which, in a general sense, approximate the paths aircraft would normally fly if path-stretching for separation purposes were not required.

The ground rules and procedures used in constructing nominal routes were as follows:

- 1. The start point for each arrival route (departure routes were not developed) was on the circumference of a circle which was centered at the primary airport and which had a radius of 55 nmi. It was found that the initial track point would normally lie inside such a circle.
- 2. There were two runway configurations in the data samples for each terminal area (see table 1), and the start point was normally the same for both configurations. The location of the start point on the circular boundary was determined by knowledge of the arrival traffic flow routings. This information was gained from (a) preferred route listings in the Airman's Information Manual, (b) observation of plotted tracks, (c) operations manuals and other facility documents, and/or (d) a priori knowledge and experience with traffic flows and procedures at the terminal of interest.
- 3. From the entry point, nominal routings normally proceeded directly to an inner (feeder) fix. It was found that, for the most part, the same feeder fixes were used for both runway configurations; however, observed exceptions were accommodated.
- 4. From feeder fix to the runway, the nominal route geometry depended upon several factors, such as weather conditions, predominate aircraft performance, the angular relationship between the feeder fix and the final approach course, and other considerations. In general, appropriate geometry from feeder fix to the runway could best be derived through repeated observations of plotted tracks. Where necessary in the design, the performance of commercial jet aircraft was assumed. It was also assumed that instrument landing system (ILS) navigation was used on the final approach, regardless of the weather (DCA "River" approach to runway 18 was an exception). However, the point of turn-on to final varied from airport to airport. On occasion, such as the parallel 14 approaches at Chicago O'Hare (ORD), different turn-on points were required between IFR and VFR conditions. When a downwind/base leg (trombone) pattern was called for, the downwind leg was constructed parallel to, and 4 to 4.5 nmi abeam of, the final approach course. The base leg in a trombone pattern permitted at 30° intercept to the final approach course at a point approximately 500 feet below the ILS glide slope.
- 5. Alternate nominal routes for light aircraft were not constructed because (a) except for DCA, relatively few were found in the sample and (b) the impact of light aircraft on excess fuel consumption was minimal. When the tracks of light aircraft deviated too far from the prescribed nominal, the track data were eliminated from delay and fuel computation.
- 6. Due to an occasional short turn-on to the final approach course, track lengths were sometimes less than nominal route lengths. Such "negative path-

stretching" was reduced through design modifications, but never was completely eliminated.

7. In order to capture the effect of local procedures on route lengths, an alternative route, referred to as "minimum approach route," was constructed for each nominal route. The minimum approach route started at the same point on the boundary circle as the corresponding nominal. From that point, the minimum approach route was constructed so as to reflect the most direct path for an ILS approach to the nearest runway in use. In this design, turn radii of jet aircraft were accommodated and, in addition, the route could not overfly the airport. Terrain, noise abatement, and the like were not taken into account in the design of minimum approach routes.

As can be seen from the "Description of Methodology" section, nominal routes were developed through an iterative process. Each configuration would be checked against a sufficient number of tracks to ensure adequate representation. Although this was primarily a judgment process, route mileage versus track length was also considered. When excessive negative path-stretching occurred, track plots were analyzed to determine whether (a) the nominal should be modified, (b) certain tracks should be dropped from further consideration, or (c) no changes should be made.

Nominal routes that were ultimately developed for the four airports in the data base are shown on figures 1 through 8. These include all routes developed except (a) VFR routes for ORD runway 14L approaches, (b) alternate routes for ORD runway 27L approaches during the dual, 32L/27R configuration, (c) Midway (MDW) routes, and (d) Fort Lauderdale (FLL) routes. Initially MDW and FLL nominals were constructed, and delay and fuel data were computed. However, due to the small number of tracks, these data were not included in the final results.

As can be seen from these nominal route configurations, many of the nominal routes nearly follow the most direct path from entry point to final approach course. This is reflected in the annotated distance data. On the other hand, minimum approach routes which differ substantially from the corresponding nominal are depicted in dashed lines. A good example is route 5B at MIA (figure 4). From track data it was found that, when landing west at MIA, aircraft through SERPA normally were vectored for an approach to 27L. The closest runway for this traffic was 27R; therefore, the minimum approach route was constructed as shown to capture the effect of these local procedures on excess fuel consumption.

To derive the amount of path-stretching delay incurred by a track, it was first necessary to find a point on the associated nominal route that closely corresponds to the start point of the track. This was accomplished in the computer program by swinging an arc through the track start point until it intersects the nominal route (see figure 9). The center of the arc was a prespecified point in the close-in pattern, normally where the minimum approach route merges with its corresponding nominal. That portion of the nominal route from the intersect point to the runway is referred to as a "comparable nominal route."

The length of the comparable nominal is called "CNOM," and the difference between track length (TRK) and CNOM is the resulting path-stretching mileage (TRK-CNOM). In the CNOM computations, adjustments were made for turning radii. Note also that when CNOM is greater than TRK, negative path-stretching mileage results. In the summary of data, negative path-stretching is treated in an algebraic manner.

HOLDING DELAY. In this study, terminal area holding delay was derived by manually recording holding times from plotted ARTS tracks. If the holding pattern was entered after the first data point of the track, then the duration of the complete holding delay was recorded. These holding data are referred to as type A holding and generally reflect holding delay that occurred after handoff from the center. If, on the other hand, the first data point of the track revealed that the aircraft was holding at that time, only the remainder of the holding delay could be recorded. Unless the time of the first data point was equal to the start time of the sample hour, it was obvious that the aircraft had been holding for an undeterminable amount of time while under control of the center. In either case, these incomplete holding delays were recorded as type B holding. Obviously, this yielded results substantially less than what actually occurred at the time, and it therefore points out the need for enroute data when deriving terminal area delay.

Figure 10 depicts examples of type A and type B holding patterns. With type A holding, the computer removes holding distance from track length by making a straight-line connection between the "time-in" point with the "time-out" point. Altitude data at these two points were also saved for subsequent use. In the case of type B holding, only "time-out" was manually extracted, and all track data prior to that point were stored as holding delay. Track length was computed from the "time-out" point to the runway. Altitude at the "time-out" point was used to represent type B holding pattern altitude.

EARLY DESCENT/SPEED CONTROL DELAY. During the analysis phase of the project it was found that (a) considerably more level-flight mileage was flown by aircraft in the data base than was needed for path-stretching delay, (b) level-flight altitudes (weighted averages) were well below 10,000 feet, and (c) level-flight mileage and speeds varied substantially from terminal to terminal. In the interest of deriving data from the FDDB relative to the profile descent program (reference 1), which was instituted after the FDDB was collected from the field, it was decided to include speed and vertical profile data in the final results.

To appreciate the delay due to early descent, consider the profile on figure 11. In that schematic, descent from 35,000 feet is initiated 27 nmi ahead of a continuous descent gradient. Assuming standard atmospheric conditions and without considering wind, the reduction in true airspeed (TAS) from 450 knots to 290 knots results in a 2-minute increase in flying time over the 27 nmi even though indicated airspeed (IAS) remains at 250 knots. This delay would have occurred in the example shown on figure 11 solely by early descent, whether or not it was the intent of the ATC system to reduce the speed of the aircraft at that point. Obviously, level flight at the lower altitude results in more fuel consumption; but this will be discussed in a later section.

Closely intertwined with early descent delay is delay caused by speed control instructions used to facilitate the traffic management function. However, a precise measurement of delay due to both forms of speed reduction requires analysis of voice recordings and additional processing of the track data that had not been provided during the programing support phase of the project. Therefore, it was necessary to manually derive an estimate of this delay from analysis of available data.

For reasons explained later, the vertical profile of each track had been computer and stored by the computer program for subsequent use in fuel consumption computation. In developing the vertical profile for arrival tracks, a regression analysis model was applied to the altitude associated with each 30-second data point. When the model indicated that the descent gradient was less than 100 ft/nmi (parameter), the track was declared as being level at that point. A typical profile is shown on figure 12. Level-flight distance and level-flight time were accumulated and stored for each track. In addition, a time-weighted altitude was computed based on the duration at each level-flight altitude. Later, data reduction programs summarized these data into altitude bands for specified sets of sample tracks. Figure 13 depicts an example of these data summaries.

Early descent mileage was derived for a given set of data by subtracting the path-stretching mileage from the level-flight mileage. In other words, these additional level miles were not needed for separation purposes and therefore could have been flown at cruise altitude. Since level-flight distances and level-flight times were available in the summary, an average track velocity was easily derived. An estimate of delay time was derived by computing the difference in flying time over the early descent miles between the time required at average track velocity and the time that would have been required at a nominal cruise speed of 450 KTAS. Obviously, in this method the effect of speed control could not be separated from the effect of early descent. Accordingly, these delay components were combined in the final results. However, additional delay occurs during descent when aircraft have been given speed instructions by the controller. An estimate of this delay was derived by computing the difference in flying time over the descent mileage (track length minus level-flight distance) between the time required at average track velocity and the time that would have been required at an average nominal speed of 265 KTAS. (At MIA, the average track velocity over all tracks in the data sample was 271 knots.) The coarseness of the foregoing estimating method is recognized. It is felt, however, that lacking a more precise method, the data so derived from the FDDB can provide important inputs to fuel conservation techniques, such as profile descent. It is expected, for example, that the point at which descent from cruise altitude is initiated, has rarely been associated with delay. Yet, by the natural phenomenon of air density, delay occurs when aircraft are descended early, whether such delay is needed or not for air traffic control purposes. Furthermore, as will be shown later, when delay is required, early descent together with speed control is a fuel-efficient way to absorb the required delay as long as speeds below clean-flap configurations are not employed. DELAY DUE TO PROCEDURAL ROUTING. In addition to holding, path-stretching vectors, and early descent/speed control, a fourth delay component, referred to as "procedural routing," was derived from the ARTS track data. These delay data relate to the added mileage embodied in the nominal route geometry as a result of local procedures. There are a wide range of factors involved in establishing procedural routings; however, it was not the intent in this study to assess the impact of the individual factors, nor to pass judgment on the procedures themselves. Rather, these data were derived solely to identify yet another area where fuel conservation procedures or techniques may have application. It will be seen, for example, that, while procedural routing average less than 1-minute delay per track, overall, there are a few routes within each terminal area which contribute to the bulk of the total procedural routing delay. Accordingly, improvements in this area to reduce excess fuel consumption would need only to concentrate on a limited number of identifiable factors.

As discussed earlier, a minimum approach route was constructed for each nominal route in order to capture the effect of local procedures on excess fuel consumption. In the basic nominal geometries (figures 1 through 8), minimum approach routes started from the same point as the corresponding nominal route. For individual track computations, however, a "comparable minimum approach route" distance computation (CMIN) was started at the track start point (see figure 9). The difference between CNOM and CMIN was the excess mileage delay attributable to local procedures (CNOM-CMIN).

EXCESS FUEL COMPUTATION. In an attempt to construct fuel flow models for the aircraft types found in the data base, it was found that actual fuel flow depends upon many factors, most of which were not available to the project. However, it was felt that good estimates could be derived from available data based on a few key assumptions. For example, aircraft performance manuals depict fuel flow in level flight a function of weight, speed, and altitude. Speed and altitude could be derived from ARTS tracks; however, it was necessary to make an assumption regarding weight for the particular aircraft type. Further, it was found, for the purpose of this project, that aircraft types could be reasonably grouped into categories in accordance with the number and type of engines and assumed weight. Table 2 shows the category grouping for the jet and turboprop aircraft used in the fuel consumption computation. These types accounted for over 99 percent of the usable FDDB arrival tracks. Another assumption involved fuel consumption during the descent phase of operation. For several reasons, it became necessary to disregard fuel consumed during descent and therefore base all findings on level-flight data. For example, excess fuel consumption attributable to path-stretching was derived from pathstretching mileage and a computed level-flight fuel flow rate (to be explained later). What this amounts to is the assumption that different descent gradients in the terminal area (after ARTS acquisition) have an insignificant effect on differences in fuel consumption.

The method for computing a level-flight fuel flow rate to be applied to delay mileage is shown on figure 14. It can be seen that the resultant rate is a weighted average based on aircraft type and the distance and speed at each level segment altitude. The application of these variables on fuel consumption

TABLE 2. AIRCRAFT TYPE CATEGORIZATION

CATEGORY	ASSUMED WEIGHT (1b)	AIRCRAFT TYPES
5	<u>-</u>	MU2, VC6, BE99, OV1, DH6, BE90, U21
6		YS11, G159, SW2, SW3, SW4, CV58, ND26, CC09, FA22, CV64, C2, FA27, FH22, C580, HP13
7		P3, C130, L188
8	_	H525, AC21, LR23, LR24, LR25, N265, T39, C500, A37
10	_	G2, FFJ
11	90,000	B737, BA11, C9, DC9
13	140,000	В727
14	220,000	B720, C135, C140
16	220,000	B707, DC8
17	320,000	DC10, L101
18	220,000	DC86
19	550,000	B747

NOTES: 1. Missing category numbers were for types not used in fuel consumption computation.

<sup>2.</sup> Assumed weights were for the arrival phase.

is shown on figure 15. A second-degree equation was developed for each of the four altitudes for the aircraft categories shown in table 2. The coefficients for the fuel flow equation were derived through application of regression analysis on the data extracted from performance manuals for the aircraft type representing each category. Two sets of coefficients were derived for commercial jet transports (categories 11 through 19). The first set, like the one shown on figure 15, represents a no-flap configuration, and the second set represents various flap settings as a function of KIAS. The flap data were obtained from the United Airlines Office, Denver, Colorado (reference 6). An example of the effect of flaps is shown on figure 16. Indicated airspeed was computed by using track velocity as KTAS and applying the following formula:

KTAS = KIAS  $\times \frac{68320 + 0.293 \text{ Z}}{6832 - 0.707 \text{ Z}}$ where Z is altitude in feet.

The weighted fuel flow rate described above was used to derive excess fuel consumption due to path-stretching delay and procedural routing. For fuel consumption during holding delay, however, a different approach was taken. It was assumed that aircraft held at optimum holding speeds and therefore fuel consumption would be in accordance with data published in performance manuals such as that shown on figure 17. These data are normally given in pounds per hour; therefore, holding duration was used in fuel computation as opposed to holding distance. In this computation a second-degree equation was used where the independent variable was altitude (Z). Regression analysis was also applied to derive holding fuel flow coefficients for each type category.

In deriving excess fuel consumption attributable to early descent/speed control delay, it was again necessary to resort to an approximation method. From analyzing fuel consumption for jet aircraft in the data base, it was found that a good average estimate of fuel consumption (lb/nmi) at cruise altitude is about half the average computed for the tracks at the weighted level-flight altitude. Therefore, after fuel attributable to path-stretching was subtracted from the total fuel consumed in level flight (634,988 lb in figure 13), the difference was divided by two. These estimates were made for only the final summaries of the total data sample for each terminal area. Also, no attempt was made to derive excess fuel consumption due to delay that accrued during descent in the terminal area as a result of speed control.

### DISCUSSION OF RESULTS

## SUMMARY OF DELAY AND EXCESS FUEL CONSUMPTION.

An overall summary of the delay data reduced from the FDDB is presented in table 3, and corresponding excess fuel consumption attributable to this delay is shown in table 4. Figure 18 depicts a bargraph comparison of the excess fuel consumption for the four airports in the sample data together with the average arrival rate over all sample hours for each airport. It is interesting to note that the average number of usable tracks in the 1-hour samples at ORD

TABLE 3. DELAY DATA REDUCED FROM ARTS TRACKS (FIELD DATA COLLECTED FOR ATC/ACAS INTERACTION STUDY)

	ORD	MIA	LAX	DCA
Total Number of Tracks	635	217	175	212
Per-Sample Average	53	18	19	18
Total Delay Time (min)	6,160	699	568	1,038
Per-Track Average	9.7	3.2	3.3	4.9
Components of Delay:				
Holding				
Number of Tracks Held Percent of Total Total Holding Time (min) Average Time Per Hold (min) Per-Track Average (min)	132 20.8 1,128 8.6 1.8	3 1.4 12 4.0 NIL	0 - - -	6 2.8 39 6.4 0.2
Path-Stretching Vectors				
Total Delay Mileage (nmi) Per-Track Average (nmi) Ratio:Delay to Nominal Route (Percent) Est. Per-Track Delay Time (min)	6,769 10.7 20 3.0	773 3.6 6 0.8	617 3.5 7 0.9	1,111 5.2 9 1.3
Early Descent/Speed Control				
(NOTE: See "Method of Approach") Est. Total Delay Time (min) Per-Track Average (min)	3,604 4.1	369 1.7	333 1.9	530 2.5
Procedural Routing				
(NOTE: See "Method of Approach") Total Delay Mileage (nmi) Per-Track Average (nmi) Per-Track Delay Time (nmi)	1,927 3.0 0.8	662 3.1 0.7	323 1.8 0.5	796 3.8 0.9

TABLE 4. EXCESS FUEL CONSUMPTION REDUCED FROM ARTS TRACKS (FIELD DATA COLLECTED FOR ATC/ACAS INTERACTION STUDY)

		•		
	ORD	MIA	LAX	DCA
Total Number of Tracks	635	217	175	212
Total Excess Fuel Burn				
All Tracks (1b)	670,009	116,493	60,424	70,995
All Tracks (gal)	100,001	17,387	9,019	10,596
Per-Track Average (1b)	1,055	537	345	335
Per-Track Average (gal)	157	80	52	50
Per-Track Cost (in dol.) @ \$.42	66	34	22	21
Excess Fuel Per Delay Component				
Holding				
TotalAll Holding (1b)	142,350	1,238	0	3,140
Per-Hold Average (1b)	1,078	413	191-19	523
Path-Stretching Vectors				
Total-All Tracks (1b)	273,954	28,644	26,498	24,692
Per-Track Average (lb)	431	132	151	116
Early Descent/Speed Control				
TotalAll Tracks (lb)	180,517	64,877	22,201	27,451
Per-Track Average (1b)	284	299	127	129
Procedural Routing				
Total-All Tracks (1b)	73,188	21,734	11,725	15,712
Per-Track Average (1b)	115	100	67	74

is about triple that at MIA, LAX, and DCA, even though the average arrival rate at ORD was only about twice that of the other airports. In general, this results from the fact that the traffic at ORD is highly regimented and could normally be associated with a nominal route; whereas, special treatment was frequently given to traffic at the other airports. In particular, during VFR weather and light or moderate traffic, local service and commuter flights at these airports would often land on a secondary runway or would otherwise be vectored such that association with a nominal route was not feasible. Further, since the FDDB samples were taken in 1-hour slices in time from the ARTS recordings, tracks at the beginning and at the end of each sample hour were also not usable for delay analysis. Overall, about 80 percent of the tracks in the ORD samples could be used in the analysis, and at MIA, LAX, and DCA about 60 percent of the tracks were suitable. Obviously, a data collection program, established to measure delay and excess fuel consumption, would require longer sampling periods than the FDDB and would avoid periods of light VFR traffic. In spite of the shortcomings of the sample data, however, the study yielded highly useful results concerning terminal area delay and excess fuel consumption. In particular, the findings from this study have direct application to efforts dealing with the development of fuel conservation procedures and techniques.

Of particular interest are the average total delay per track (table 3) and the average excess fuel consumption attributed to that delay (table 4). Note that, while individual delay components appear relatively small, the aggregate of these data can be considered substantial, particularly at ORD. For example, the average time in system, from ARTS acquisition to landing, was about 21.5 minutes for the 635 tracks; however, using methods previously described, nearly 45 percent of that time (9.7 minutes) was calculated to be delay. The average nominal route distance for these tracks was 54 nmi, so that without holding, path-stretching, or speed control, the average time in the system would be slightly over 14 minutes. Therefore, even discounting procedural routing delay and the added effect of early descent, a delay of 7 to 8 minutes remains. In addition to the effect on fuel consumption, absorbing this delay prior to ARTS acquisition would have the added effect of reducing the simultaneous number of aircraft under approach control by seven to nine aircraft. Obviously, the impact on center workload would depend on how and where the delay was absorded by ARTCC.

In connection with excess fuel consumption, the average aggregate amount of 1,055 lb (157 gallons) per track at ORD appears substantial in light of fuel costs. To properly view these data in connection with the national posture on energy conservation, it is necessary to extend the estimates of excess fuel consumption to a longer time frame, such as to an annual basis. It was felt, however, that the sample data used in this project were not sufficiently representative to make a statistically valid annual extrapolation. Without belaboring the point, it seems obvious that estimates of this nature require especially designed data collection criteria. Nevertheless, to sense the order of magnitude of what the average values on table 4 project to on a annual basis, one can apply simple arithmetic to the ORD data. It was found, for example, that the present demand/capacity ratio at ORD is about the same as it was when the data used in this project were collected (1974-75). Also, inspection of the schedules in the Official Airline Guide indicates that the

demand is at a nearly continuous peak from 9 a.m. to 9 p.m. Therefore, a first-order estimate of excess fuel consumed daily by arrival aircraft during the 12 peak hours at ORD can reasonably be derived by extending the 100,000 gallons of excess fuel from the sample data (table 4) in the following manner. First, since each sample hour yielded about 48 minutes (80 percent) of usable track data, it is necessary to divide 100,000 by 0.8 to extend the 12 sample hours to 12 complete hours of operation. This computation yields an estimate of 125,000 gallons of excess fuel consumption for 12 peak hours. Assuming the sample data are sufficiently representative of the average operations at ORD during peak hours, an annual estimate of excess fuel consumption due to delays to arrival aircraft ranges from 32.5 million gallons (based on a 5-day week) to 39 million gallons (based on a 6-day week). Extending the total delay time for the ORD sample (6,160 minutes, table 3) in the same manner yields an annual estimate of from 2 to 2.4 million minutes of delay. Using an average fuel cost of 42¢ per gallon and an average direct operating cost without fuel of \$10.00 per minute (reference 7, B727, extrapolated to December 1978), a range in cost to the users due to arrival delay at ORD is estimated to be from 33 to 40 million dollars, annually. It should be emphasized at this point that, when comparing these delay cost estimates with other estimates, the data from the this study did not include terminal area delay that might have accrued prior to the ARTS track acquisition point (i.e., ARTCC holding vectoring and speed control). Also, no attempt was made to derive delay and excess fuel consumption for the departure or ground operation phases of operations. Obviously, an analysis of departure tracks and of tracks in the close-in enroute area would be a natural extension of the work conducted in this study, and, together, these analyses could provide good estimates of total terminal area delay costs.

In addition to the foregoing, it would also be of interest to derive annual estimates of delay costs at other terminal areas. Such estimates were not made in this study for MIA, LAX, and DCA for two reasons. First, it was felt that the sample data did not sufficiently represent the wide range of conditions which cause delay at those airports (i.e., prolonged periods of IFR weather and/or saturated traffic conditions, etc.). Second, traffic at these airports has not reached the near steady state conditions of the 9 a.m. to 9 p.m. traffic at ORD. Therefore, an estimate of annual delay costs would require an analysis of the traffic conditions that occur throughout the year. Such an analysis was beyond the scope of this study; therefore, no attempt was made to extend the MIA, LAX, and DCA data beyond that shown in tables 3 and 4.

Concerning the delay and excess fuel consumption problem, one further point is in order. It was found in this study that a representative commercial jet (B727) would burn about the same amount of excess fuel as the average of the ORD tracks (1,055 lb) if all delay (9.7 minutes) were absorbed while holding at 20,000 feet. This does not support the argument for the postulated fuel savings of rigid profile descent procedures where all terminal area delay is absorbed in holding patterns at metering fixes. It does, however, support work already started, and encourages new work relative to fuel-efficient methods for consuming delay.

#### COMPONENTS OF DELAY AND EXCESS FUEL CONSUMPTION.

In order to provide a better understanding of the impact of fuel consumption of the various delay-absorbing methods, it was decided to partition the delay data from the ARTS tracks into the four components shown in table 3. The contribution each component makes to total delay and total excess fuel consumption at each airport in the data base is shown in table 5. Salient aspects of these data are included in the detailed discussion of each delay component that follows.

TABLE 5. RATIO OF DELAY COMPONENTS TO TOTAL DELAY AND TOTAL EXCESS FUEL CONSUMPTION

	O	KD.	M	IA	L	AX	De	CA
DELAY COMPONENT	% OF TOTAL DELAY	% OF EXCESS FUEL						
Holding	18.6	21.2	NIL	1.1	0	0	3.8	4.4
Path-Stretching	30.9	40.9	25.0	24.6	27.3	43.9	26.5	34.8
Early Descent/Speed Control	42.3	26.9	53.1	55.7	57.6	36.7	51.0	38.7
Procedural Routing	8.2	10.9	21.8	18.7	15.2	19.4	18.4	22.1

HOLDING. Of the four delay components shown in table 5, "holding" provides the best index of the demand versus capacity relationship. However, to measure demand and the holding that results when demand exceeds capacity, it is necessary to analyze data well in advance of the point where the tracks are acquired by ARTS. Since only ARTS track data were available to this study, no estimate of terminal area demand was attempted, and the holding delay shown in table 3 consists only of the holding times that could be extracted from these tracks (see Method of Approach). In spite of this limitation, the holding delay derived from the ORD tracks is considered significant. For example, even though the arrival rate averaged 66 aircraft per hour (figure 18), about 1 in every 5 aircraft encountered holding delay. This is a good indication of the excess in demand over capacity during the sample periods and, judging by recent traffic statistics, is probably indicative of the present operation at ORD. (Obviously, this excludes the triplearrival runway operation which has recently been introduced during suitable periods of wind, weather, and runway braking conditions.)

It is interesting to note that the ratio of aircraft that were held (20.8 percent) is about the same as the ratio of total holding fuel to the total excess fuel consumed (21.2 percent) as shown in table 5. This is due to the large amount that each hold costs in excess fuel (1,078 lb, on the average).

Regarding the holding time at ORD, it should be remembered from the Method of Approach that when aircraft had been held by the center, only the time subsequent to ARTS acquisition could be tabulated for this study. Of the 132 aircraft that were held, 87 were of this type, averaging 6.7 minutes of holding time after entry into the ARTS system. Tracks of the remaining 45 aircraft that were held were acquired prior to holding start time, which generally indicates that approach control had instructed these aircraft to hold. These tracks had an average hold duration of 12 minutes. Although one might assume that the latter duration provides a good estimate of average holding time for all aircraft that were instructed to hold (either by ARTCC or by approach control), there was nothing in the data to verify this assumption. Therefore, a more valid approach in estimating holding delay cost would be to consider system holding time as opposed to peraircraft holding and to separate the enroute data from the terminal data. While estimates of enroute holding delay data require separate study, estimates for the ORD terminal data can be made by extending the holding delay in table 3 in a manner analogous to the approach taken for total delay, above. First, recalling that the 12 sample hours amount to about 9 actual hours of track data, then the 1,128 minutes of holding delay computes to an average of 125 minutes per peak hour, or 1,500 minutes for 12 consecutive hours of peak traffic (normal ORD operations from 9 a.m. to 9 p.m.). Simple arithmetic yields a first-order annual estimate of terminal area holding at ORD of from 390,000 minutes (for 5-day week) to 468,000 minutes (for 6-day week). Extending the excess fuel consumption at ORD due to holding delay from table 4 in a similar manner produces an annual estimate ranging from 7.4 to 8.8 million gallons.

It should be noted that the data used in this study were collected prior to the implementation of profile descent procedures. The intent of these procedures is to eliminate holding and other delay at low altitudes inside the metering fix. The fact remains, however, that when aircraft are put into holding stacks, demand on the airport has exceeded the effective capacity of the airport for some undefined period of time. With an equal demand/capacity ratio, the application of profile descent alone merely shifts the holding delay to a higher altitude. Obviously, the fuel savings by holding at higher altitudes depends upon several factors, including type and weight of the aircraft, holding speed, flap configuration, etc. An example of the effect of holding altitude is presented in table 6. For the fuel consumption of the track data, it was assumed that all aircraft held in a clean configuration. This yielded an average holding fuel flow rate of 7,917 lb/hr for the five categories shown. This rate is slightly lower (2.3 percent) than the weighted average at 10,000 feet from the UAL data (8,101 lb/hr). The difference is due, in part, to the lower holding altitudes (overall average of about 9,000 feet), and also to the no-flap assumption for the heavy aircraft. With the same distribution of holding times by type of aircraft, a weighted average of 7,105 lb/hr was computed from the United Airlines (UAL) data for holding at 20,000 feet. This is 10.3 percent less than the track average, and 12.3 percent less than the weighted average at 10,000 feet from the UAL data. For holding times of 10 minutes per hold, these differences yield 135 and 166 lb less fuel. respectively. At current fuel prices (42¢ per gallon), it is estimated that an average savings of 8 to 10 dollars for each 10-minute hold would result by

TABLE 6. HOLDING FUEL FLOW DATA

FUEL FLOW DATA FROM UNITED AIRLINES (UAL)

HOLDING FUEL CONSUMPTION FROM SAMPLE TRACKS (99.2% OF HOLDING FUEL)

A/C TYPE	FUEL FLO	W (1b/hr) 200 KIAS)	CATEGORY	HOLDING FUEL BURN (1)	TIME	RATIO OF HOLDING TIME (PERCENT)
(WEIGHT)	10,000 ft	20,000 ft		2012. (1)	,, (	(I Little)
B737 (90K 1b)	4,185 (clean)	3,865 (clean)	11	21,547	272	25.4%
B727-200 (140K 1b.)	7,085 (clean)	6,616 (clean)	. 13	61,014	483	45.1%
DC8-61 (220K 1b)	11,600 (10° flaps)	9,520 (clean)	16	32,714	195	18.2%
DC10 (320K 1b)	13,880 (slats)	11,060 (clean)	17	20,494	103	9.6%
B747 (550K 1b)	24,500 (5° flaps)	21,200 (1° flaps)	19	5,409	17	1.6%
Weighted Average*	8,101	7,105	Total Average	141,178 Rate 7,917	1,070 lb/hr**	

<sup>\*</sup> UAL fuel flow rates weighted by proportion of the total holding time the representative category in the sample data was held.

<sup>\*\*</sup> Average holding altitude at ORD was approximately 9,000 feet.

holding at 20,000 feet as opposed to the holding altitudes in the ORD sample data. By applying the difference in final consumption due to holding altitude to the annual estimate made earlier, a difference ranging from 0.8 to 1 million gallons of fuel results.

The absence of any significant amount of holding at the other airports in the sample data is probably more reflective of the periods during which the data were collected than anything else. However, it can be assumed that the requirement for prior reservation for landing at DCA minimizes holding delay at that airport, except possibly during prolonged periods of IFR weather. All DCA data used in this study were collected during periods of VFR weather. Although 3 hours of the LAX data were during IFR weather, the traffic demand during these periods did not exceed the airport capacity to any noticeable degree. This was generally the case throughout the LAX sample data; therefore, it is felt that the zero holding delay as well as all other delay measures at LAX are not representative of that busy airport. At MIA, the weather, traffic demand, and multiple runway operation, together, militate against the need for holding delay. It is expected that only during seasonal periods of peak itinerant traffic will holding delay of any substantial amount be required at MIA.

PATH-STRETCHING VECTORS. As with holding, the delay due to path-stretching vectors at ORD clearly stands out over the other airports. Because of the near steady state traffic demand at ORD from 9 a.m. to 9 p.m., it is felt that the ORD data are representative of the path-stretching delay generally encountered by arrival aircraft during these 12 peak hours each day in the years when the data were collected. Data from the other airports, on the other hand, are not considered so representative, since many factors which cause delay were not a part of the data collection criteria for the ATC/ACAS study.

To more clearly perceive the significance of the ORD data, consider the fact that the average time in the system was just over 21 minutes; viz, from track acquisition to touchdown, not including holding. In this interval of time, aircraft, on the average, flew about 20 percent (10.7 nmi) further than the nominal route distance and consumed about 431 lb (64 gallons) of excess fuel. Extending these data to form annual estimates by the method previously described yields a range of 2.2 to 2.6 million delay miles, annually, and a range in excess fuel consumption of 13 to 16 million gallons. Further, the average fuel burn rate over all path-stretching delays at ORD was slightly over 40 lb/nmi or about 8,660 lb/hr. This is approximately 22 percent more than the weighted holding fuel burn rate at 20,000 feet (7,105 lb/hr) shown in table 6. Therefore, extending this difference to the annual estimate yields a savings of between 2.4 and 2.9 million gallons of fuel in favor of high-altitude holding.

It should be pointed out that this finding is not in contraposition with the discussion under "total delay," since the latter included the impact of early descent and speed control. As will be shown, these delay-absorbing techniques can be highly fuel-efficient. The difference in fuel-efficiency between

path-stretching and early descent/speed control can be seen from the data in table 5 for ORD, LAX, and DCA. At these airports, the percentage of excess fuel attributed to path-stretching is considerably higher than the corresponding percentage of delay absorbed, while the reverse is true in the early descent/speed control data. The incongruity in the MIA data is explained in the next section.

EARLY DESCENT/SPEED CONTROL. As described in the "Method of Approach," delay and excess fuel consumption attributable to early descent and speed control were derived by an estimation method due to the lack of complete information about the tracks. Regardless of this limitation, these data are relevant to the objectives of the study because the findings show that (a) unintentional delay and excess fuel consumption can result from early descent and (b) when delay is required, early descent together with speed control are fuel-efficient techniques for absorbing the required delay, as long as speed below clean-flap configuration are not employed. Further, it will also be shown that the use of speeds that require flaps is not a fuel-efficient method of absorbing delay.

The effectiveness of the early descent/speed control method of absorbing delay can be seen from the data in table 5. Except for MIA, the delay ratio is considerably higher than the excess fuel ratio. Further, at ORD, LAX, and DCA, the delay fuel burn rate due to early descent/speed control computes to 4,156, 4,010, and 3,096 lb/hr, respectively. The fuel-efficiency of this method becomes obvious when these rates are compared with the "holding" fuel burn rates in table 6. In the MIA data, a reverse trend was exhibited. It was found, however, that the average track velocity at MIA was 271 knots. This is 6 knots higher than the nominal terminal area speed used to compute speed control delay. Therefore, no delay was attributed to controller speed control instructions, leaving all delay attributed to early descent. It is not known whether such delay was intended or not, but, if intended, it would have been more fuel efficient to keep the aircraft longer at cruise altitude applying speed control at altitude and during descent, as required.

The impact of altitude and speed control strategies at the four airports is further exhibited by the level-flight data in table 7. In these data, the differences in strategies between MIA and LAX are clearly evident. At LAX, for example, the level-flight distance not used for path-stretching vectors was less than 8 nmi, on the average; whereas, at MIA, it was more than 19 nmi. Also the average IAS at LAX was estimated to be 208 knots, which is close to minimum clear-flap configuration speed. At MIA, the computed IAS was 242 knots, just under the FAA speed limit below 10,000 feet. The impact of the differences in operational procedures is reflected in the level-flight fuel consumption. At MIA, the average fuel consumption during level flight was computed to be 729 lb per track, while at LAX the average was 405 lb. It is felt that this difference of 324 lb per track is highly significant, particularly since the average total delay (table 3) was about the same at the two airports. It appears also that during the data collection periods of the FDDB, the LAX operations were fairly close to profile descent procedures even though these procedures were actually established by FAA at a later date.

TABLE 7. LEVEL-FLIGHT DATA

LEVEL FLIGHT DATA (PER-TRACK AVERAGES)	ORD	MIA	LAX	DCA
Level Flight Distance (nmi)	27.5	22.7	11.4	21.7
Path-Stretch (nmi)	10.7	3.6	3.5	5.2
Early Descent Distance (nmi)	16.8	19.1	7.9	16.5
Ratio of Level Distance to Track Length	42%	37%	22%	35%
Weighted Level-Flight Altitude (ft)	7,000	7,500	8,500	8,000
Track Velocity (knots)	214	271	236	244
Computed Indicated Airspeed* (knots)	193	242	208	216
Time in Level Flight (min)	7.7	5.0	2.9	5.3
Fuel Burn in Level Flight (1b)	999	. 729	405	375
Fuel Burn Rate (lb/nmi)	36.3	32.1	35.5	17.3

<sup>\*</sup> IAS was computed from track velocity (TAS) at the weighted level flight altitude with the following equation:

IAS = TAS 
$$\times \frac{68320 - 0.707 \text{ (alt. in ft)}}{68320 + 0.293 \text{ (alt. in ft)}}$$

At ORD and DCA, the level-flight distances not needed for path-stretching vectors were about the same, on the average (16.8 and 16.5 nmi, respectively). However, there is a marked difference in the average fuel consumption during level flight (999 lb versus 375 lb), which results from (a) differences in aircraft-type distribution and (b) differences in speed control application. The effect of aircraft-type distribution is shown on figures 19 and 20. In figure 19, it can be seen that over all the FDDB samples the ratio of excess fuel consumed by the B727 to total excess fuel consumption is about the same as the percentage of B727's in the data base. At DCA, on the other hand, B727's consumed about 63 percent of the excess fuel even though this type constituted only about 38 percent of the aircraft in the sample data (figure 20).

In connection with speed control it can be seen from table 7 that, on the average, aircraft speeds at DCA (216 KIAS) were above clean-flap configuration. At ORD, however, an average IAS of 193 knots indicates that speeds requiring flaps were frequently issued by the controllers. The effect of flaps on fuel flow rate is shown on figure 16. This effect can be more clearly demonstated by comparing the fuel required to absorb 1 minute of delay through either speed control, vectoring, or holding. If the example aircraft were indicating 210 knots at 5,000 feet (276 KTAS), it would take 23 nmi for a reduction to 180 KIAS (194 KTAS) to absorb 1 minute of delay. Over this distance, 1,104 1b would be consumed at 210 KIAS, and 1,564 lb at 180 KIAS with 15° flaps; therefore, the 1-minute delay costs 460 lb of fuel by this speed reduction. Using path-stretching vectors at 210 KIAS, it would take 3.8 nmi to absorb 1 minute of delay, and the additional fuel consumption would be 182 lb. Thus, the excess fuel for the path-stretching delay is about 40 percent that of speed control when 15° flaps were required. The holding fuel flow rate for the example aircraft at 20,000 feet is about 9,520 lb/hr (table 6), or 159 lb/min. This is about 12 percent less than the path-stretching example, and 65 percent less than speed reduction to 180 KIAS. However, now consider the case where speed is reduced from 250 KIAS (269 KTAS) to 210 KIAS to absorb 1 minute of delay for the example aircraft. For these speed differences, 23.5 nmi are required to absorb 1 minute of delay. Fuel consumption over this distance is 1,128 lb at 210 KIAS and 1,010 lb at 250 KIAS; therefore, the speed control delay would cost 118 1b of fuel. One minute of path-stretching delay at 250 KIAS would cover a distance of 4.5 nmi and would require 193 lb of fuel. In this case, the delay fuel cost using speed control is about 60 percent of that using pathstretching vectors and about 75 percent of the "holding" fuel consumption (159 lb). To summarize the above, estimated fuel consumption for the example aircraft in figure 16 to absorb 1 minute delay with different ATC strategies are:

- 1. 118 1b--Speed reduction (250 to 210 KIAS)\*
- 2. 159 1b--Holding at 20,000 feet
- 3. 182 1b--Path-stretching vectors (210 KIAS)\*
- 4. 193 1b--Path-stretching vectors (250 KIAS)\*
- 460 1b--Speed reduction (210 to 180 KIAS)\*\*

<sup>\*</sup> No flaps \*\* 15° flaps

These data were based on level flight in the terminal area. It is evident that reduced speed in the descent phase (both terminal and enroute) is more efficient than any of the level-flight strategies, since the differences in fuel flow during descent are small for all operationally acceptable speeds. An example of this efficiency can be seen by comparing the data in reference 5 (DC8, 220,000 lb) for the long-range descent (0.78M/250 KIAS) with the high-speed descent (0.83 M/340 KIAS) data. From a final cruise altitude of 35,000 feet, the long-range descent requires 134 nmi and burns 1,440 lb of fuel. Starting from the same point and altitude the high-speed scenario would require 30 nmi of level flight and 104 nmi for descent. Estimated fuel consumption would be 690 lb for level flight and 1,080 lb for descent, yielding a total of 1,770 lb. From the common point, the flying time for the long-range descent is about 3 minutes more than for the high-speed descent profile. Therefore, if ETA's were based on the high-speed descent, the long-range descent profile could be used to effect 3 minutes of delay, while, at the same time, saving 330 1b of fuel. Since these data are based on the 250 KIAS FAA speed limit below 10,000 feet for both profiles, time, fuel, and distance, differences accrue prior to reaching 10,000 feet. On the assumed profile, the 10,000-foot point is 32 nmi from touchdown. The decision point for selecting the longrange descent would therefore be 102 nmi from that point and about 72 nmi from where the high-speed descent profile reaches 20,000 feet. Obviously, many factors enter into that decision, including the accuracy in estimating flying times and delay requirements. Since examination of these factors is not part of this study, the only point that can be made here is the fuel efficiency of the early descent (30 nmi, prior to high-speed profile) integrated with reduced speed. In this example, fuel was saved whether the delay was needed or not. However, early descent which causes level flight at lower altitudes is wasteful of fuel if the corresponding delay is not required.

PROCEDURAL ROUTING. As discussed in the "Method of Approach," a "minimum approach route" was constructed to correspond with each "nominal route" in order to assess the impact of local procedures on delay and excess fuel consumption. These minimum approach routes define the shortest path from the track start point to the closest runway in use (without overflying the airport), taking into account aircraft performance and instrument approach requirements. Procedural routing delay is defined as the difference in flying time over the nominal route as compared with the corresponding minimum approach route. Excess fuel consumption attributed to local procedures was derived by applying the weighted fuel flow rate in level flight to the difference in these route lengths.

As can be seen from table 3, the overall average delay attributed to local procedures was less than 1 minute per track for all airports in the data base. The average excess fuel consumption due to this delay component (table 4) was also a modest amount, varying from 67 to 115 1b per track. It is felt, however, that in this case average values are misleading insofar as providing the kind of information needed for fuel conservation methodology. It can be seen from figures 1 through 8 that many of the nominal routes provide a fairly direct path from the terminal entry point to the close-in approach pattern. In each configuration, however, there are a few cases

where the minimum approach route is substantially shorter than the corresponding nominal route. While it was not within the scope of this project to analyze the reasons for these procedural routings, it is important to identify the impact on excess fuel consumption.

Table 8 depicts excess route mileage and excess fuel consumption due to procedural routing for those nominal routes where the average nominal route mileage was 3 or more nmi longer than the average minimum approach route distance. In these data, only nominal routes with 10 or more tracks are included. At ORD, there were 26 nominal routes constructed to match the runway configurations and other conditions in the data base. Of these, only the five shown in table 8 had an average procedural routing delay greater than 3 nmi. The 186 tracks assigned to these routes (29.3 percent of the ORD sample) had an average delay of 7.3 nmi due to procedural routing, which resulted in an average excess fuel consumption of 290 lb per track. Also, these tracks accounted for 73.4 percent of the total excess fuel consumption in the ORD data that was attributed to procedural routing delay. In actuality, route 1B (PAPI to 27R) and route 7A (VAINS to 14R) together accounted for about 54 percent of the ORD procedure routing total while accommodating about 19 percent of the sample traffic (121 tracks). By inspecting these route geometries in figures 1 and 2, it appears that a considerable amount of airspace is reserved for departure traffic or other ATC purposes. Whatever the reasons may be, it would appear that a concerted effort to conserve fuel would involve a close examination of the ORD procedures with a view toward more direct routing of the heavy arrival traffic from the northeast when landing west and also from the southwest when landing to the southeast.

At MIA, 3 of the 11 nominal routes accounted for about 75 percent of the excess fuel consumption attributed to procedural routing. About 35 percent of the MIA data base tracks were assigned to these three routes. When landing west, route 1A traffic (through PINKS) was routed considerably east of the most direct route (perhaps to avoid FLL airspace), resulting in procedural delay of 7 nmi and excess fuel consumption of 258 lb, on the average. When landing west, traffic from the northwest (route 6B through NEWER proceeded to the MIA VORTAC before turning to intercept the downwind leg. This resulted in 171 lb of excess fuel for the 6 extra nmi. Also in the west configuration, traffic off of V-35 through SERPA normally vectored south of the airport to land on runway 27L. These tracks averaged 5.9 nmi more than the minimum approach route (north of the airport) and consumed 195 lb of additional fuel. It is assumed that this procedure resulted from the manner in which traffic is distributed between the "North" and "South" arrival controllers.

In the LAX data, 2 of the 12 nominal routes accounted for 89 percent of the excess fuel consumption attributed to procedural routing. The 62 tracks assigned to those routes (35.4 percent of the LAX tracks) were all from the northwest sector, whether landing west (route 5A) or landing east (route 6B). Rationale for these routings was not apparent from the data; however, noise abatement procedures and terrain problems are well known factors in the LAX area.

TABLE 8. EXCESS MILEAGE AND FUEL CONSUMPTION DUE TO PROCEDURAL ROUTING

AIRPORT- NOMINAL ROUTE	AVERAGE EFM-P PER TRK (nm1)	NO. OF SAMPLE TRACKS-	NO. OF TRACKS ON NOMINAL	PERCENT OF "N" (Z)	EFB-P IN SAMPLE (1b)	EFB-P ON NOMINAL (1b)	PERCENT OF SAMPLE EFB-P (%)	AVERAGE EFB-P PER TRK (1b)
ORD-		635			73,188			
1B	7.8		64	10.1		18,798	25.7	293
6B	10.4		18	2.8		5,335	7.3	296
7A	7.9		57	9.0		20,332	27.8	356
8B	5.4		25	3.9		5,149	7.0	205
8A	3.6		22	3.5		4,381	6.0	199
TOTAL	-		186	29.3		53,995	73.4	-
AVERAGE	7.3							290
MIA-		217			21,734			
lA	7.0		31	14.3		8,005	36.8	258
5B	5.9		19	8.8		3,715	17.1	195
6B	6.0		27	12.4		4,622	21.3	171
TOTAL	-		77	35.5		16,342	75.2	
AVERAGE	6.4							212
LAX-	,	175			11,725			
6B	5.0		41	23.4		7,287	62.1	177
5A	3.6		21	12.0		3,147	26.8	149
TOTAL			62	35.4		10,434	89.0	11 - 11 - K
AVERAGE	4.5							168
DCA-		212			15,712			
LA	3.6		32	15.1		1,930	12.3	60
18	3.3		32	15.1		2,315	14.7	72
3В	3.6		21	9.9		833	5.1	39
4B	11.7		13	6.1		2,936	18.7	225
5A	5.0		36	17.0		4,349	27.7	120
TOTAL	-		134	63.2		12,363	78.7	
AVERAGE	4.7							92

Legend: EFM-P - Excess miles due to procedural routing EFB-P - Excess fuel burn due to procedural routing

The five nominal routes listed in table 8 for DCA accommodated 63.2 percent of the DCA traffic in the data base and accounted for about 79 percent of the excess fuel consumption due to procedural routing. Of particular note are routes 4B and 5A (Ironsides to runway 18 and Gilby to runway 36, respectively), where 23 percent of the DCA traffic accounted for over 45 percent of the excess fuel consumption attributed to local procedures. In both cases it appears that traffic is vectored off the most direct route to avoid the airspace used for departure traffic exiting through the Casanova VORTAC. Again, aircraft-type distribution of the DCA traffic renders a less severe impact on fuel consumption due to local procedures than at the other three airports. The fuel flow rate of these aircraft averaged just under 20 lb/nmi as opposed to 33 to 40 lb/nmi at MIA, LAX, and ORD.

# FACTORS CONTRIBUTING TO DELAY AND EXCESS FUEL CONSUMPTION.

GENERAL. From the previous discussion, it is evident that no single formula fits the four airports in the data base relative to delay and excess fuel consumption. This can be seen more clearly by reference to figure 21, where excess fuel attributed to the four delay components is presented for each airport as a percentage of the total excess fuel consumption derived for all tracks in the data base. It is fairly obvious that the ORD data in figure 21 are indicative of the persistently heavy traffic demand at that airport. For example, in the total FDDB, the ORD samples accounted for 51 percent of the tracks, while the other three airports together produced the other 49 percent. Also, the excess fuel consumption derived for the ORD portion of the tracks amounted to 73 percent of the total for the entire data base. It can be inferred, from the excess fuel attributed to each component of delay (excluding procedural routing), that demand exceeded capacity throughout most of the sample from ORD, while such was not the case at MIA, LAX, and DCA. The one apparent incongruity to this inference is the ratio of the early descent/speed control data depicted for MIA. However, as explained earlier, the MIA tracks were generally descended early, but very little speed control was exercised. Although it cannot be proven with certainty from the data, it appears that the practice of descending aircraft early at MIA resulted in unintentional delay and unperceived excess fuel consumption. Note, however, that these data were collected before profile descent procedures were implemented. Such procedures should minimize results of this type.

In the LAX data, the converse of MIA seems evident. Generally, the LAX tracks conformed to a fuel-efficient profile when delay was not required. In this regard, however, the author has some reservation with respect to how nearly the sample data reflect true demand at either LAX or MIA. Although not apparent from figure 21, the excess fuel consumption data at DCA were strongly influenced by aircraft-type distribution and the requirement for landing slot reservations. In addition, the FDDB data collected from DCA were recorded during good VFR weather conditions, and frequently arrivals were assigned to a secondary runway. During IFR conditions, a single arrival runway is used at DCA. It is expected that such conditions would produce substantially different results for DCA than those derived from the FDDB.

In order to identify the effects of weather, runway configuration, and other factors on excess fuel consumption, it is necessary to apply different sampling criteria than those used for the ATC/ACAS study. However, in an attempt to extract as much information as possible from the available FDDB collected for that study, it was decided to organize the ORD data into the various groupings as shown in figures 22 through 25 for comparison purposes. These data are discussed in the subsequent sections.

EFFECT OF WEATHER. In the ORD sample, 3 data hours were collected during IFR weather when the parallel 14L/14R runway configuration was in use, and 3 hours were collected during VFR conditions with the same runway configuration. Figure 22 depicts average delay time and excess fuel consumption per track for these two conditions. In these data, delay during IFR exceeded delay during VFR by 19 percent (12.3 versus 10.3 min), and excess fuel consumption in the IFR data was 26 percent greater than in the VFR data (1,415 versus 1,118 lb per track). Although, these results may appear to be as expected, there was considerable variation among the sample hours as shown in figure 23. Note, for example, the holding data from example I1, as compared to samples I2 and I3, and in sample V1 as compared to V2 and V3. Also note that the average vector delay in sample V3 (14.7 nmi) was almost twice that in sample V2 (7.9 nmi). These differences do not seem to correlate with the landing rates shown at the top of figure 23, nor can they be explained from other aspects of the ORD data base. Evidently, such differences are a result of other factors, such as short-term demand, controller strategy, differing center/tower procedures, etc., which require more information to isolate than was available in the FDDB.

EFFECT OF RUNWAY CONFIGURATION. From discussions with ORD personnel, it was found that the 32L/27R, or "dual," configuration was considered to be the most efficient runway configuration for arrival aircraft (departures use 32R/27L). This seems to be confirmed by the sample data on figure 22, where delay with the dual configuration (7.7 min) was about 25 percent less than the average delay with the parallel, 14L/14R configuration (10.3 min) where both operations were during VFR weather. However, as shown on figure 23, inconsistencies between the dual-sample data also exist. Note the average delay mileage in the D2 sample (22.7 nmi) as compared with 6.6 and 8.8 nmi in D1 and D3, respectively. Also, the average delay in the V2 sample (8.5 nmi) is about the same as the better dual samples. Again, the reasons for such variation between samples were not detectable from available data.

EFFECT OF APPROACH PATTERN GEOMETRY. (Note: Programing support for the project was canceled before it was decided to analyze delay and excess fuel consumption due to early descent and speed control. Consequently, these data could not be extracted for all data groupings, such as those shown in figures 24 and 25).

Nominal route geometry was classified as being either a straight-in, a base leg, or a trombone (downwing/base leg) pattern. These geometries provide differing degrees of controllability and therefore require different control strategies to produce the required spacing in the arrival sequence. Accordingly,

it is of interest to determine the results of such strategies on delay and excess fuel consumption.

As shown in figure 24, aircraft that flew a trombone pattern (most controllability) had the most delay (25.5 nmi), while the straight-ins (least controllability) had the least delay (12.6 nmi). Average delay for tracks on a base leg pattern (23.0 nmi) also seems to correlate with controllability; i.e., slightly less than trombone and substantially more than straight-in. This relationship between delay and controllability is a more or less natural characteristic of a radar (ATC) environment, where, generally, "the end justifies the means." It is unfortunate, however, that speed control data were not available for these comparisons, since aircraft on a straight-in are more likely to be given lower speeds than aircraft on the other patterns. As shown earlier, speeds requiring flaps have a pronounced effect on fuel consumption relative to the amount of delay absorbed. Also, the matter of "holding" by ARTCC should be considered. Feeder (holding) fixes on a straightin pattern are normally closer to the final approach gate than the feeder fixes on base leg and trombone patterns. This may very well influence the decision by the center to put aircraft in a holding pattern, and, when held, the shorter distance could affect the handoff time from ARTCC relative to other arrivals. Whether or not this is true could not be ascertained from the FDDB, since ARTCC holding data were not available.

EFFECT OF ENTRY SECTOR. The data on figure 25 were organized in order to see if a relationship exists between excess fuel consumption and terminal entry sector. From the 12 sample hours at ORD, there does appear to be a definite relationship between these factors. Note that the average excess fuel consumption for the 193 tracks through the southeast (SE) sector was about half that for the 135 tracks through the southwest (SW) sector. Part of this difference can be attributed to the percentage of straight-ins from the SE (43 percent) versus a nearly equal ratio (51 percent ) of base leg patterns from the SW.

More noticeable, however, in the differences in holding data (321 versus 48 lb) and in procedural routing (215 versus 40 lb). Actually, "holding" and procedural routing combine to account for most of the differences between all entry sectors. From visual inspection of arrival and departure tracks in the ORD data, it appears that the small amount of procedural delay in the SE sector, as compared to other sectors, is a direct result of the way the airspace is segregated between arrivals and departures. However, there is no explanation from the data regarding the reduced amount of holding in the SE sector. Possibly because the traffic flow is somewhat heavier, there may be a different arrangement between approach control and the terminal sector in the ARTCC.

OTHER FACTORS AND CONSIDERATIONS. Undoubtedly there are numerous other factors that should be considered relative to excess fuel consumption. Furthermore, most factors are closely interrelated. However, to identify all factors, and their interrelationship would require a data collection and analysis effort far beyond the scope of this project. It is felt, however, that the discussion in this section, together with the discussion in the

previous section, should impart pertinent information relative to the consumption of excess fuel in terminal areas. These data were reduced from real-world tracks with the only motivation being to extract and disseminate maximum knowledge relative to this most important problem.

### DESCRIPTION OF METHODOLOGY

#### GENERAL.

It was pointed out in the "Method of Approach" section that data recorded by ARTS require a considerable amount of processing and editing before being effectively applied to operational analyses. It was found that in order to derive credible delay data from ARTS tracks it is necessary to provide for manual intervention at various points in the process. Figure 26 depicts a simplified block diagram of the methodology developed to derive delay and excess fuel consumption in the terminal area. Subsequent sections briefly describe each functional block.

Before proceeding with a description of the process, it should be pointed out that the cornerstone of the approach taken was the use of nominal routes against which track data were compared. While other methods for deriving delay, such as a relative frequency distribution of flying times, could have been applied, it was felt that a direct, one-to-one comparison of track versus nominal route yielded the most accurate and complete information regarding path-stretching delay, controller strategy, procedures, and other factors.

It can be seen from figure 26 that a considerable amount of work preceded the development of nominal routes. On the surface, this may appear overdone. As it turns out, however, very small deviations can cause substantial differences in the final results, particularly when dealing with high-density terminal area traffic.

# TRACK DATA BASE PREPARATION.

For the most part, the data-recording capability was established at selected ARTS facilities to monitor system performance and assist in maintenance and modification needs. The data tapes are retained for a 15-day period for legal purposes and following that period may normally be obtained from the facility, provided appropriate coordination and administrative procedures are followed. However, since the recording of ARTS data are not directed towards analysis of the ATC system, considerable effort is required in the selection and preparation of data elements needed to meet specific analytical objectives. Figure 27 depicts an overview of the data preparation steps followed during the ATC/ACAS Interaction Study which resulted in the FDDB used in this delay and fuel consumption analysis. A detailed discription of the software and other data preparation activities may be found in reference 3 and associated program documentation. Although this data preparation process was designed specifically for the ATC/ACAS work, it is probably

representative of the effort required to prepare field data for most analytical applications. Accordingly, a brief description of the principal program functions follows.

CONV 79 (Block 1.1). This is a straightforward conversion of data on seventrack ARTS tapes to nine-track tapes compatible with the National Aviation Facility Experimental Center (NAFEC) computers.

TLP (Block 1.2). This "Track Listing Program" performs the following functions:

- a. Selects tracks for a specified sample hour from the source data.
- b. Converts the ARTS position coordinates to a coordinate system common to all locations in the FDDB.
- c. Provides a listing of tracks in the sample hour together with data describing the quality of each track.
- d. Produces an output tape containing selected elements of information needed for succeding steps.

DATSYN (Block 1.3). This program has two primary functions:

- a. Adds altitude information to tracks which do not have mode C transponder data. On the ATC/ACAS project, altitude data were taken from pilot/controller voice tapes and encoded for input to computer program DATSYN.
- b. Performs editing of track data to eliminate anomalies and spurious data which can normally be expected in field-derived data. Some of the abnormalities can be screened by program logic, while others require manual inspection and evaluation of the track data. For example, in the ATC/ACAS project an altitude change rate criterion was used wherein the program could detect most of the spurious altitude data. On the other hand, gaps in the track positional data required manual evaluation and input editing commands for the program to make the necessary track modifications.

DATA TRANSLATION PROGRAM (DTP) (BLOCK 1.4). The primary functions of this program are:

- a. Performs parabolic (nine-point) smoothing of track position and altitude which renders the data more suitable for fine-grain analysis than results from the ARTS alpha-beta smoothing algorithm.
- b. Performs between-point interpolation to produce track data points at 1-second intervals. This was necessary for the ATC/ACAS project, since ACAS logic was predicated on a 3-second cycle time; whereas, ARTS data are acquired at approximately 4-second intervals (antenna rotation rate of 15 revolutions per minute (rpm)).
- c. Produces an output tape of smoothed, 1-second "snapshot" data of all tracks in the system. It was the set of DTP output tapes that provided the data source for this study.

# TRACK SIMPLIFICATION AND PLOTTING.

At the outset, it was apparent that plots of the arrival tracks would be required in the development of nominal routes and, later on, in associating individual tracks with the appropriate nominal routes for path-stretching computation. In the interest of efficiency in plotting and other processing, it was desirable to (a) reformat DTP data from interleaved scan form to chronological track history form and (b) reduce the number of data points that defined each track. The programs shown on figure 28 were developed to reformat, simplify, and provide visual presentation of the FDDB tracks.

HALFTRACK (Block 2.1). Due to the age of the DTP tapes, numerous read errors occurred. Also, for some locations, a sample hour required two DTP tapes. The HALFTRACK Program was written to copy the DTP data onto one tape by eliminating odd-second data points. Also most read errors were eliminated.

TRACKS (Block 2.2). This program converts interleaved scan data into a chronological track history format. The choice of formats depends upon project requirements. The ATC/ACAS project was interested in the instantaneous relationship of one aircraft to another; whereas, this project needed entire track histories for delay and fuel computations.

SIMTRACK (Block 2.3). This program performs the following range of functions to facilitate manual and computer-based analysis of track data:

- a. Eliminates overflight tracks from the data base.
- b. Eliminates arrival and departure tracks with track durations less than specified values.
- c. Detects and flags data gaps and computes an estimate of distance flown during the gap in the track data.
- d. Reduces the number of data points in the track history by redefining straight-line segments with intervals of 30 seconds (parameter) between data points. A linear regression technique was used to determine straight-line track segments. When the error sum of squares value exceeded a specified value, the track was assumed to be turning. When this occurs intervals of 4-second spacing are retained in the track history.
- e. SIMTRACK also computes an estimate of the aircraft's final track heading and velocity. Normally, these data have little meaning for departures; however, for arrivals the data can be used to determine landing runway, an estimate of landing time, and other uses depending upon project requirements.

AREAPLOT (Block 2.4). The plotting program developed for the project was designed to satisfy a wide range of requirements. This includes plotting combinations of routes, fixes, runways, track histories, and other data. In view of the magnitude of the data base and the fact that all tracks had to be plotted one or more times, the program was designed so that a complete tape of track histories could be plotted in a single operation of the program. This

is accomplished by fitting six, 20 x 20 inch x, y grids into the basic CALCOMP grid and plotting up to eight (option) tracks on each grid. When data for the six grids have been plotted, the program stops, allowing the operator to position new paper on the plotter bed and then restart the program. This continues until all tracks on the tape have been plotted. Some of the more important options of the AREAPLOT program are:

- a. Information to be plotted, i.e., arrival tracks, departure tracks, nominal routes, reference fixes, boundary circle, etc. Also, tracks to be plotted may be selected by aircraft identification or, if not selected, all tracks on the tape will be plotted, with an option of up to eight tracks on each x, y grid.
- b. Plotting scale 6 nautical miles (nmi) per inch was used in this project.
- c. Color coding i.e., routes and background data of one color and three tracks each of a different color enhanced the readability for this project.
- d. Fix identification may or may not be plotted, as desired. If plotted, the height of the fix identification (ID) lettering can be controlled separately from other lettering.
- e. Real time associated with track position may be plotted with control over height of the plotted numbers.
- f. The center and radius of a boundary circle can be controlled. For this terminal area work, a circle with 55-nmi radius centered at the primary airport was used.
- g. Compass roses, strategically located at various points on the grid, may be plotted to the desired size. This facilitates the measurement of bearing data.

From the list of user options, it is obvious the AREAPLOT program provided a key interface between the analyst and the computer. As will be seen, this program box will appear as an integral part of most of the steps in the methodology developed for the project. Its description at this point is for continuity purposes only.

### AIRWAY DATA BASE PREPARATION.

As shown on figure 26, the preparation of airway data and terminal area geometry can be performed in parallel with the preparation of track data, leading up to the development of nominal routes. In a general sense, the purpose of the system of programs depicted on figure 29 is (a) to establish readily usable disk files of real world data which are available from different sources, and (b) to provide a convenient method to extract data needed for specific project requirements.

Through use of this system, three interrelated files are created on the Sigma 8 disk pack for ready access. The programs which process the FAA Airport Master tapes (blocks 3.1, 3.2, and 3.3) produce an alphabetized file of airports with three-letter identification where all extraneous (i.e., administrative) data have been removed. Also, three programs (blocks 3.4, 3.5, and 3.6) are used to provide a similar file of navigation aid (NAVAID) data from the FAA NAVAID Master tape. The EXPER program (block 3.7) creates a file of airway, route, and associated fix data from the Controllers Chart Supplement Subscriber tape established and maintained by National Ocean Survey, National Oceanographic Atmospheric Administration (NOAA). The AIREDIT program (block 3.12) provides the capability to enter manual corrections to the airway/route file, based on the diagnostics provided by the EXPER program. Manual corrections to the airway/route data base are made on the tape input to AIREDIT through use of the TRANSFORM 3 program (block 3.10) together with the AIRWAY 3 program (block 3.9). AIRPULL (block 3.8) is an extractor program which selects, from the disk files, airway and route data contained within a specified lat/long box of up to eight sides (convex polygon). The OMNIPLOT program (block 3.11) plots, under a wide range of options, the airway/route data selected by AIRPULL.

Several of the programs shown on figure 29 (asterisked) were developed during a previous area navigation high-altitude network study. Detailed descriptioning these programs are contained in reference 4. In particular, programs TRANSFORM 3 and AIRWAY 3 are network design oriented where route or airway design is primarily a manual function. Through the use of simple command codes, design decisions can be transformed into an airway/route structure data base which is amenable to further computer processing required for effective network design. Further, these programs together with AREAPLOT provided the essential software used in this project for nominal route development.

## TERMINAL AREA GEOMETRY EXTRACTION.

Once the airway/route data have been stored on disk files, the software shown on figure 30 can then be used to select and plot data for the particular terminal area of interest. In this project, the selected data provided the starting point for nominal route development, discussed in the next section. The CALCOMP plots served as initial worksheets where the plotted data could be cross-referenced with preferred route descriptions, prestored flight plan data, operating manuals, letters of agreement, etc. The programs shown on figure 30 were briefly discussed in previous sections of this report.

#### NOMINAL ROUTE DEVELOPMENT.

As discussed in the "Method of Approach" section, nominal routes were developed in order to derive estimates of path-stretching delay. A schematic of the steps taken in the development process is presented in figure 31. The programs shown on figure 31 have been discussed in earlier sections and the ground rules and procedures used in constructing nominal routes were presented in the "Method of Approach."

From previous RNAV work, it was recognized that development of route structures is a highly judgmental process which can only be automated to a limited degree. Therefore, as can be seen from figure 31, nominal route development was centered around a manual design effort, assisted, to the extent practicable, by computer techniques. The development effort proceeded in an iterative fashion so that configurations could be checked against representative track data and modified as necessary. This process has broad applications for studies dealing with terminal area traffic control.

# ASSOCIATION OF TRACKS WITH NOMINAL ROUTES.

Following the development of nominal routes, the next step in the process was to associate each track with the appropriate nominal route for subsequent path-stretching delay computation. This step is depicted in figure 32. Although program logic could have been developed which would make the correct association most of the time, considerable manual review would still be required due to the many vagaries in the track data. Therefore, it was decided to leave track association as a manual function. In addition to associating tracks with nominal routes, other additions and/or changes to the track history data base were required. Manually derived data were encoded for input to the TRAMP program which interpreted the command functions and made the appropriate additions and changes to the data base. The following input commands were used in this project.

ASSIGN TRACK (AT). With this command, the program stored the assigned nominal route in the track file and automatically put in runway coordinates as the last track position (for arrivals).

DELETE TRACK (DT). If the track was considered inadequate for mileage computation, the DT command was used. Track data were not deleted from the file with the DT command; however, a flag was set indicating that the track was not to be included in mileage computation. Holding data, on the other hand, could be retained and processed for deleted tracks (see "HOLD," below).

CHANGE STATUS (CS). For various reasons the "arrival, departure, overflight" status code was sometimes in error in the ARTS data. When the error was detected from the plotted track, the CS command was used to correct the data base.

HOLD. This command was used to input "start" and "end" holding times as observed on the track plot. The program computed estimated distance flown during these times and stored holding times, distance flown, and holding altitudes in the data base. Track histories were also modified so that holding distances were not included in track mileage, since holding data were treated as a separate delay component. Holding data were retained for deleted tracks as well as for tracks associated with a nominal route. In this way, hourly holding data could be tabulated for the terminal area.

MODIFY TRACK (MT). Due to the fact that the FDDB samples were 1-kour slices in time from the ARTS recorded data, track data at the beginning and at the end of each sample were frequently of insufficient duration for delay analysis.

In lieu of deleting all incomplete tracks, however, it was found that many tracks could be salvaged through additions and/or slight modifications to the track history. In this process, the overriding consideration was not to impose bias in track distance. If this could not be done, then the track would be deleted from the mileage computation. In addition to incomplete tracks, there were other reasons why track history data required modification. As mentioned earlier, holding data needed to be extracted. Also, on occasion, the plots would show that the aircraft landed on a different runway than the one in the corresponding nominal route. To avoid bias in the distance computation, it was necessary to replace the runway of the nominal route with the actual runway. Bias could also be introduced if the beginning portion of the track was too far from the nominal. This section of the track could easily be truncated by use of the MT command.

Several input commands for TRAMP were programed to aid in the modification process. These commands, used in conjunction with the MT command, were as follows:

- a. New Fix (NF). This command established an identifiable position which can subsequently be used to insert, through MT, additional points in the track history in order to make an incomplete track usable for mileage computation. This method of track modification was only used if it could be judged from the available track data that the actual track would have had to pass in close proximity to the added points. A good example of this application was where the last few data points of the track history indicated that the aircraft had made the turn onto base leg just before the end of the sample hour. In all probability, a normal approach with no further pathstretching was made from that point on. By adding one or more track "fixes," track mileage could be computed without bias.
- b. Take Nominal From (Fix) to (Fix). Another way of inserting posttional data in track history was to use portions of the assigned nominal route as track position data. Again, precautions were taken with the use of this command, so as not to introduce bias in the track length computation.
- c. Take To, From, or From/To (Times). With these three separate commands, track histories could be modified through use of the real time data associated with track position. Generally these commands were used for extracting holding data or to remove positional data at the beginning of a track which would cause bias in mileage comparisons.

The inputs to the TRAMP program were encoded in free format which generally resembled a high-order computer language. Diagnostics were provided to detect input errors, and plots of the associated track histories were made for manual review. Corrections were encoded and processed in the same manner as the original data.

### ROUTE MILEAGE AND DELAY COMPUTATION.

At the completion of the track association function, a data base of usable tracks was available for route mileage and delay computation (block 7.0, figure 26). As shown on figure 33, this function consists of two computer programs (NOMLEN and TRKDAT), plus some manual work which was added during the analysis phase of the project. The rationale for manually derived early descent and speed control data (block 7.3) was discussed in the "Method of Approach" section.

The NOMLEN program (block 7.1) computed route lengths for nominal and minimum approach routes which served as an input to the TRKDAT program. In this computation, adjustments were made at each turn in the route to account for aircraft turn radius.

The TRKDAT program performed a range of functions necessary to derive delay due to path-stretching, holding, local procedures, early descent, and speed control. The principal functions of TRKDAT are embodied in the discussion of these delage components in the "Method of Approach" section.

## EXCESS FUEL COMPUTATION AND DATA SUMMARY.

The final step in the process to derive estimates of excess fuel consumption in the terminal area from ARTS tracks is shown on figure 34. This process consists of the FUELBURN program, which computed excess fuel consumption due to excess mileage and holding, and the SUMMARY 3 program, which provided a wide range of higher order summaries of these data. In addition, a manual effort was added during the analysis phase of the project to derive an estimate of excess fuel consumption due to early descent. Details of excess fuel computation are presented in the "Method of Approach" section.

As shown on figure 34, the SUMMARY 3 program provides the capability to produce higher level summaries in accordance with a range of input options. These higher level summaries for the ORD data in the FDDB are included as an appendix to this report.

#### CONCLUSIONS AND RECOMMENDATIONS

Based on the results of this study, it is concluded that:

- 1. Analysis of ARTS track data is an effective method to derive credible estimates of terminal area delay and excess fuel consumption. However, delay to arrivals may start to accrue while aircraft are still under center control. Therefore, for more complete analysis of terminal area delay, data should also be collected from appropriate sectors of the ARTCC.
- 2. To avoid misleading results, the process used to derive delay estimates from track data should provide for manual intervention at appropriate points. This follows from the fact that the recorded data are characterized by anomalies, spurious data, and other vagaries.
- 3. The need for fuel-efficient, delay-absorbing techniques is clearly evident from the delay and excess fuel consumption data derived in this project for the Chicago O'Hare (ORD) airport. From these data, it was estimated that annual delay costs to ORD arrivals is in the range of 33 to 40 million dollars. Although ORD represents a "worst case" situation at the present time, delay at other major hubs is rapidly increasing due to the accelerating traffic growth.
- 4. Rigid procedures that absorb all terminal area delay in high-altitude holding stacks do not provide the most fuel-efficient way to absorb delay. Fuel-efficient scenarious to absorb delay should include appropriate combinations of the following strategies, which are listed in descending order of fuel efficiency:
- a. Reduced speed while descending from enroute altitude to metering fix altitude,
  - b. Reduced speed at enroute cruising altitude,
- c. Descent to an appropriate lower cruising altitude to effect further speed reduction,
- d. Reduced speed between the metering fix and the approach gate, but not to speeds requiring flaps,
  - e. High-altitude holding.
  - f. Path-stretching vectors, and
  - g. Lower speeds near the approach gate for fine-grain control.

From the conclusions, it is recommended that:

1. The methodology developed for this project be expanded and/or modified, as necessary, to provide a general purpose capability for the analysis of terminal area operations from the data being recorded by ARTS.

- 2. Procedures and techniques be developed which incorporate the delay-absorbing strategies listed in conclusion number 4.
- 3. A data collection program be established to measure the efficiencies of the delay-absorbing strategies after implementation.

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   June 1974.

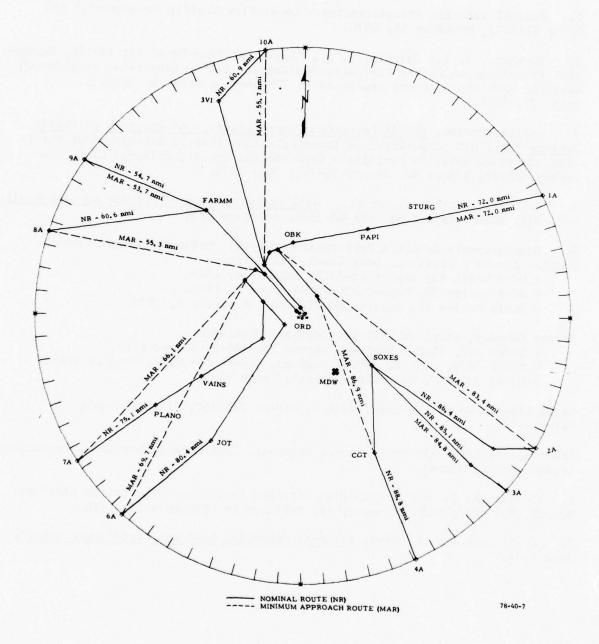


FIGURE 1. NOMINAL ROUTES--ORD CONFIGURATION A (PARALLEL OPERATIONS--RUNWAYS 14L/14R)

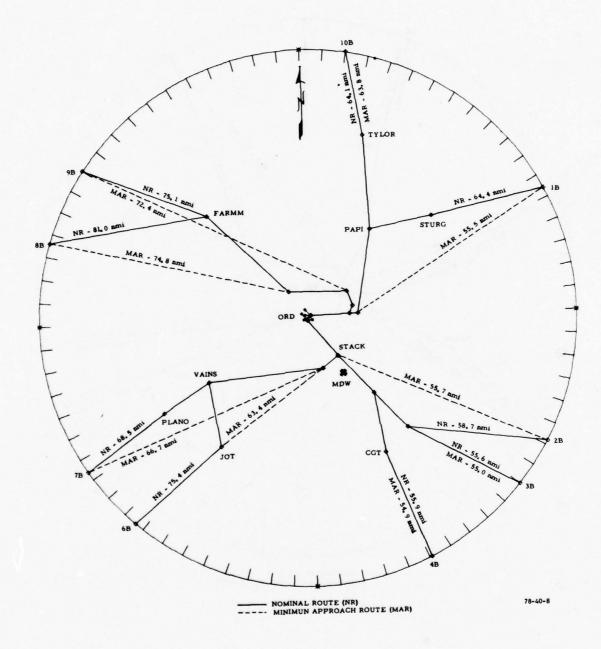


FIGURE 2. NOMINAL ROUTES-ORD CONFIGURATION B
(DUAL OPERATIONS-RUNWAYS 32L/27R)

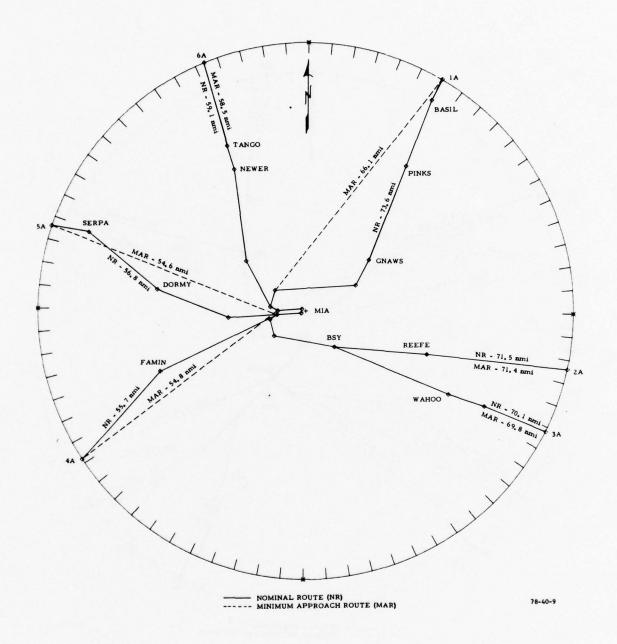


FIGURE 3. NOMINAL ROUTES-MIA CONFIGURATION A (EAST OPERATIONS-RUNWAYS 9L/9R)

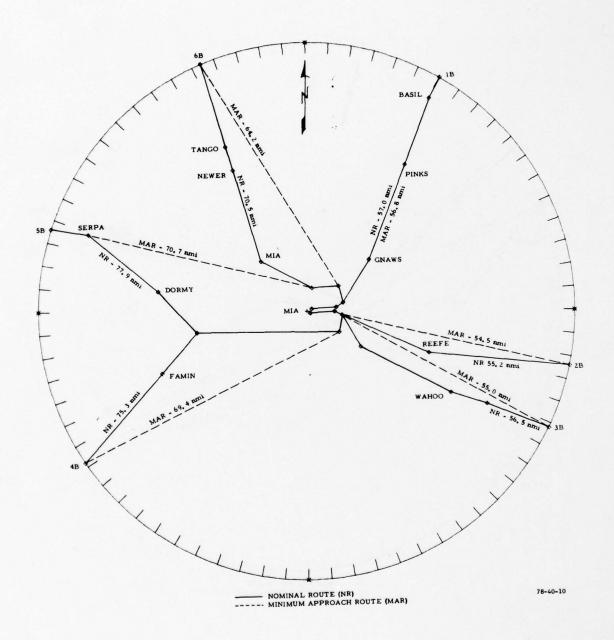


FIGURE 4. NOMINAL ROUTES--MIA CONFIGURATION B
(WEST OPERATIONS--RUNWAYS 27L/27R)

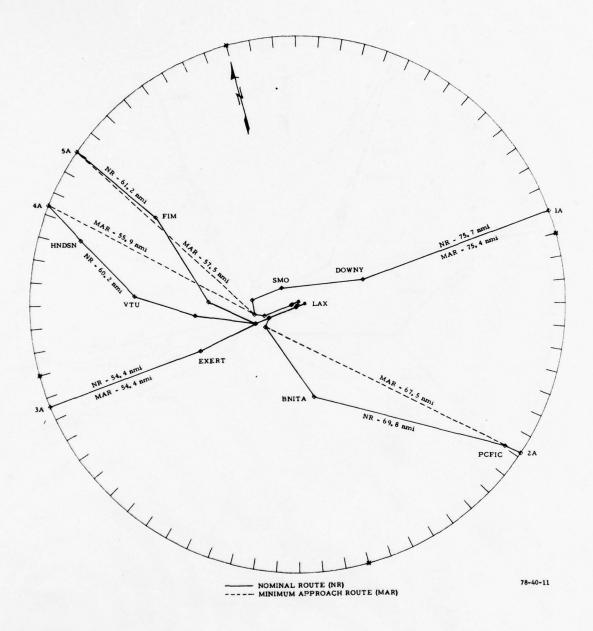


FIGURE 5. NOMINAL ROUTES--LAX CONFIGURATION A (EAST OPERATIONS--RUNWAYS 6/7)

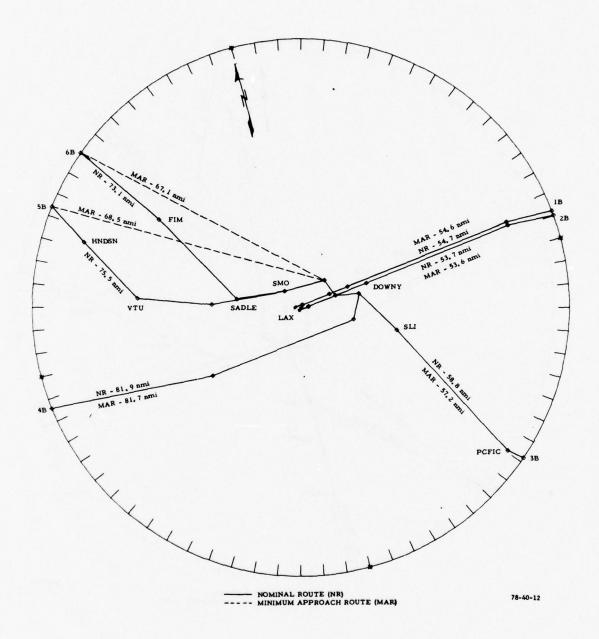


FIGURE 6. NOMINAL ROUTES--LAX CONFIGURATION B
(WEST OPERATIONS--RUNWAYS 24/25)

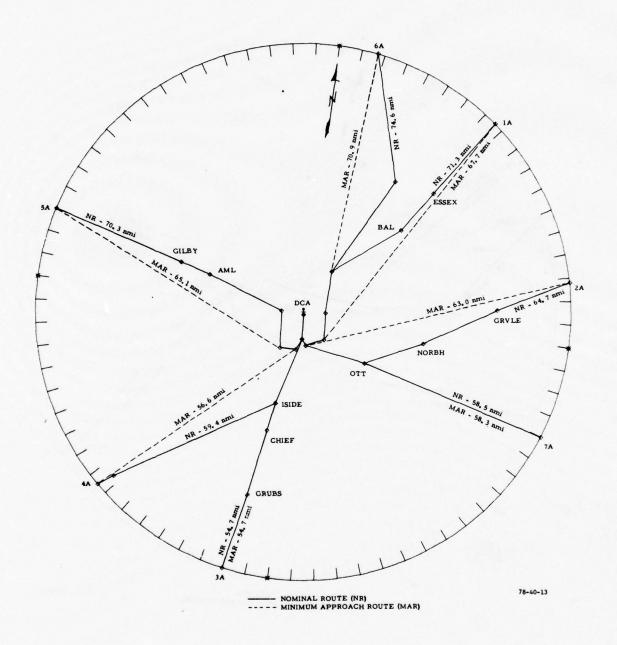


FIGURE 7. NOMINAL ROUTES--DCA CONFIGURATION A (NORTH OPERATIONS--RUNWAY 36)

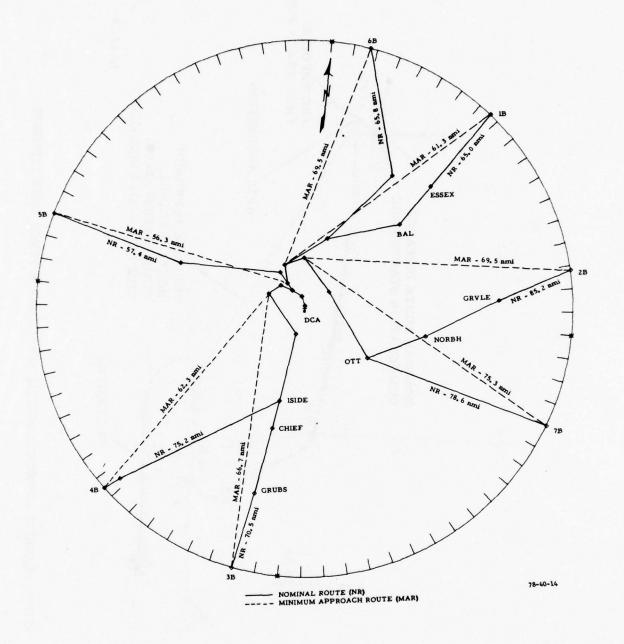
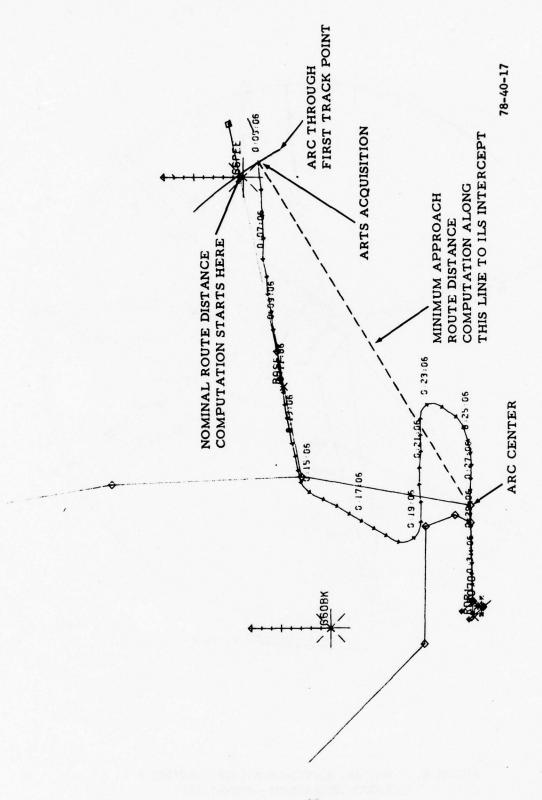


FIGURE 8. NOMINAL ROUTES--DCA CONFIGURATION B
(SOUTH OPERATIONS--RUNWAY 18)



METHOD TO DERIVE COMPARABLE NOMINAL AND MINIMUM APPROACH ROUTE DISTANCES FIGURE 9.

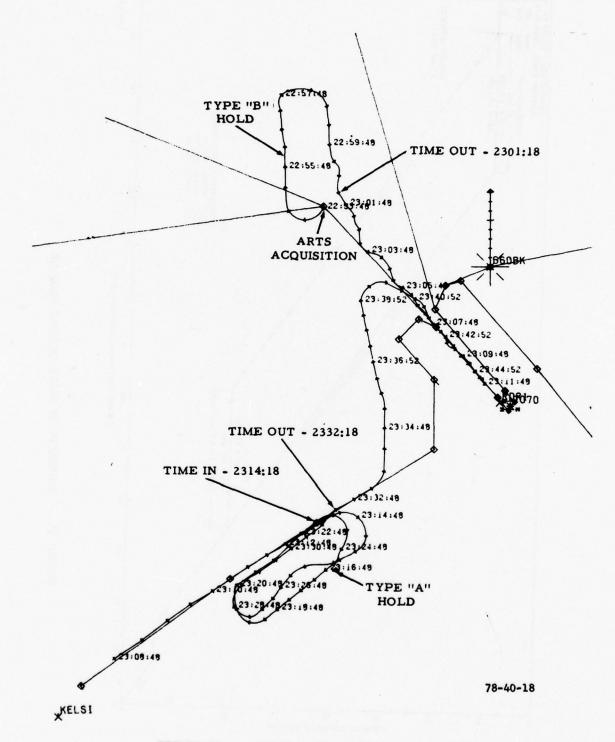


FIGURE 10. EXAMPLE OF HOLDING DATA EXTRACTION

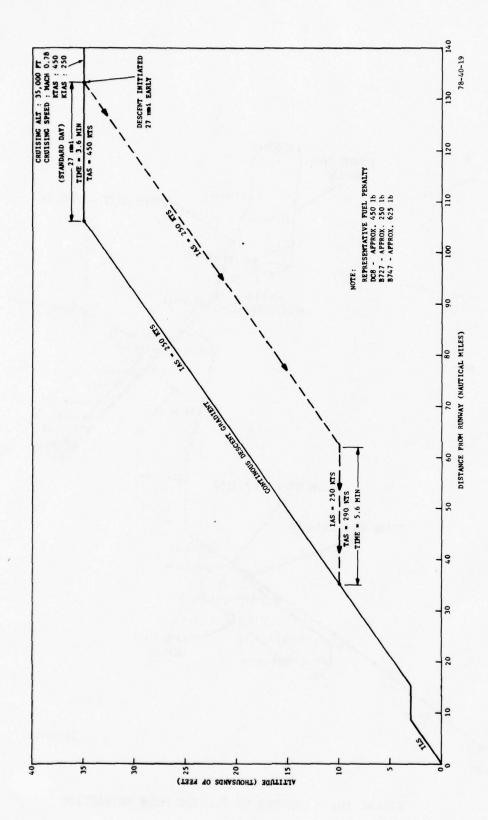


FIGURE 11. SCHEMATIC OF EARLY DESCENT PROFILE

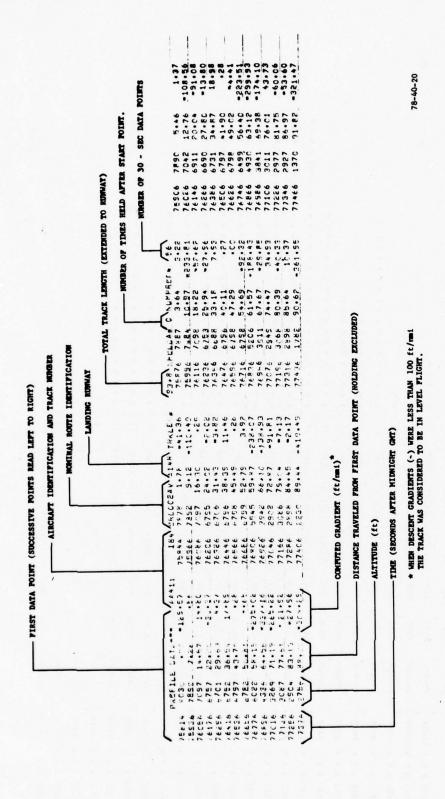
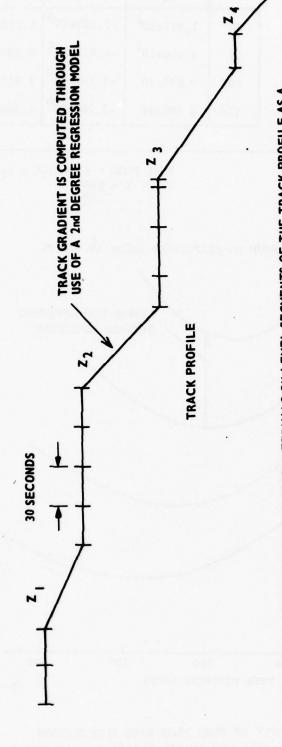


FIGURE 12. TYPICAL VERTICAL PROFILE DATA

ALTITLCE BANG	Ę.	AVERAGE TIPE	Les	2	TETAL	es	Les ICCUNTI									
99595 \$ x > 11499	24.756	E: + 2	619	123.750	28:30	3608	60	:	:			-				
115CC 3 x > 10999	14.613	2:30	265	37.840	7:30	797	6	:								
11CCC 3 x > 1C499	20.625	•:•	501	41.250	08:6	1003	(4									
1080C 3 x > 9999	10.256	3:46	669	316.360	104:14	11821	17	:	•	:			:	1.		
10000 1 × > 9499	21.874	5:13	742	196.870	00:10	6681	5		:							
98CC 3 x > 8999	10.555	80:4	531	182.590	45:38	.850	=	:		:	:					
860 X X 2006	18.465	.:10	*	351.290	119:20	14151	13	:	:			:	•	:		
8ECC 2 x > 7999	28.477	6:38	616	1053.660	+05:50	33789	37	:	:	•		:		:	:	:
8CCC 3 x > 2499	28.865	7:01	852	1382.830	541:56	\$68-4		:	:		:	:	:	:	:	:
75CC 2 x > 6999	20.265	7:03	821	2442.630	1056:36	76390	93		:			:		•		
1000 2 x > 6499	20.761	7:81	973	2222.860	1010:52	111-8	(F)		:			:				
6666 X X 2009	34.857	9:18	1179	2562.820 1205:34	1205:34	91998	78	:	:		:	:		:	:	
6665 < x & 2009	34.073	10:15	1195	3066.540	1523:40 1	107597	96			•			:	•		:
EECC 3 x > 4999	33.025	10:04	1388	1618-900	813:58	25389	6.4					•		•		
8000 3 x > 4499	28.38+	16:8	1322	879.910	434:50	41007	31		:					•		:
45CC 3 x > 3999	20.764	£:37	558	394.510	205:58	18208	15									
460C \$ X > 3+99	11.438	*:05	511	125.820	**:26	5627	=		•		:					
35CC 3 x > 2999	10.27	£:38	825	308-130	147:16	15682	15		:	:	:	:	:			
3000 3 x > 2499	126.61	1:24	1050	175.200	59:28	11557	11		•				1			
25CC 3 x > 1999	1330	-	0	000.		0	υ									
2000 3 x > 1499	- 333•		0	2000		В	o									
15CC 3 x > C	1000		0	3000		0	0				-		1			
	10000		-	-												

FIGURE 13. LEVEL-FLIGHT DATA SUMMARY



I. FUEL BURN IS COMPUTED AT 30-SECOND INTERVALS ON LEVEL SEGMENTS OF THE TRACK PROFILE AS A FUNCTION OF (a) AIRCRAFT TYPE, (b) ALTITUDE, (c) SPEED. GRADIENTS OF LESS THAN 100 FT. PER NAUTICAL MILE ARE CONSIDERED LEVEL.

2. FUEL BURN AND LEVEL DISTANCE ARE SUMMED FOR ALL LEVEL SEGMENTS.

3. A WEIGHTED FUEL FLOW RATE IS DERIVED BY DIVIDING TOTAL FUEL BURN BY TOTAL LEVEL DISTANCES.

4. THE WEIGHTED FUEL FLOW RATE (Ib/nmi) IS APPLIED TO EXCESS MILEAGE TO DERIVE EXCESS FUEL CONSUMPTION.

78-40-23

FIGURE 14. METHOD FOR DERIVING A WEIGHTED FUEL FLOW RATE

	A <sub>0</sub>	A <sub>1</sub>	A <sub>2</sub>
SL	1.511×10 <sup>2</sup>	-7.209×10 <sup>0</sup>	1.270x10 <sup>-1</sup>
5K	1.160x10 <sup>2</sup>	-4.822x10 <sup>0</sup>	8.002×10 <sup>-2</sup>
10K	9.836x10	-3.722×10 <sup>0</sup>	5.813x10 <sup>-2</sup>
15K	8.604×10	-3.002×10 <sup>0</sup>	4.368x10 <sup>-2</sup>

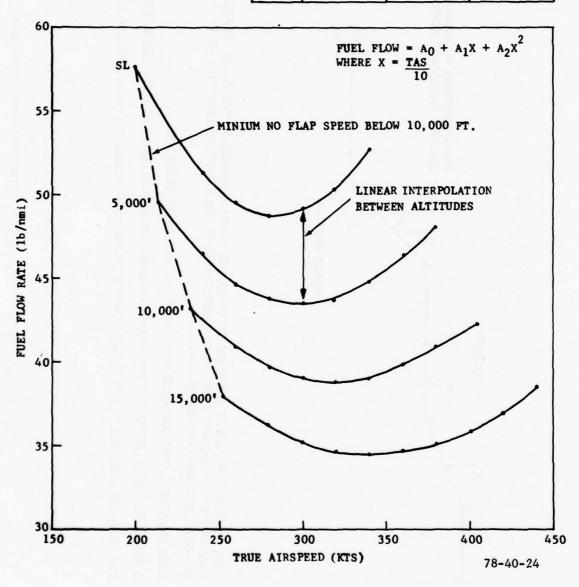


FIGURE 15. EXAMPLE OF FUEL FLOW RATE COMPUTATION (DC8--220,000 1b--NO FLAPS)

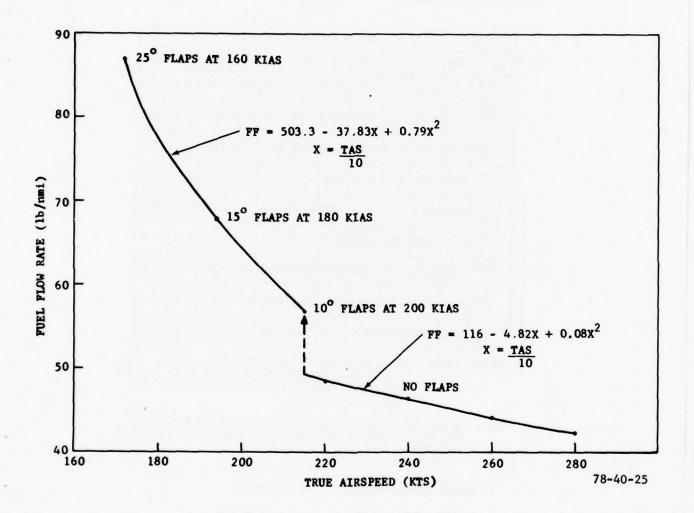


FIGURE 16. EFFECT OF FLAPS ON FUEL FLOW RATE (DC8--220,000 1b--5,000 ft)



HOLDING PLANNING 2 ENGINES 2 AIRBLEEDS

FUEL FLOW BASED ON ISA ADJUST FUEL FLOW ± 1% PER ± 5°C\_ISA DEVIATION TOTAL FUEL FLOW - LB/HR

	1
WEI	GHT

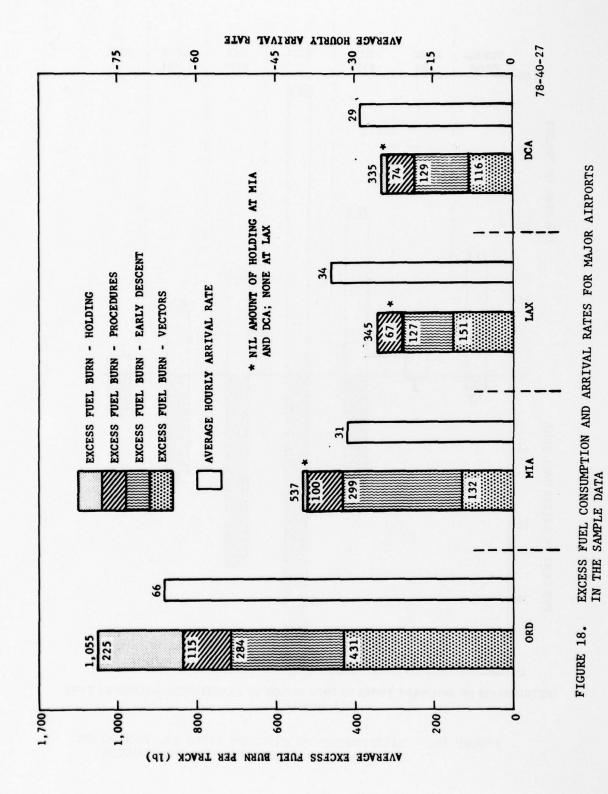
							7					
	PRESS				GROS	S WEI	GHT -	- 1000	) LB			
	ALT-FT ISA-°C	115	110	105	100	95	90	85	80	75	70	65
	35000 -54		5480	5120	4790	4480	4250	4030	3830	3640	3470	3310
	30000 -44	5340	5070	4810	4620	4390	4190	3990	3820	3660	3500	3360
	25000 -35	5270	5070	4850	4620	4410	4230	4050	3890	3730	3580	3450
	20000 -25	5290	5070	4850	4670	4500	4330	4170	4020	3880	3740	3620
	15000 -15	5440	5230	5020	4810	4640	4480	4320	4170	4030	3900	3770
	10000	5590	5420	5220	5000	4820	4660	4510	4360	4220	4090	3990
2	5000 5	5780	5600	5400	5200	5030	4870	4720	4590	4460	4330	4200
	1500 12	5940	5760	5550	5340	5170	5040	4890	4740	4610	4490	4360
	S.L. 15	6030	5830	5620	5430	5270	5110	4960	4820	<b>4690</b>	4560	4420

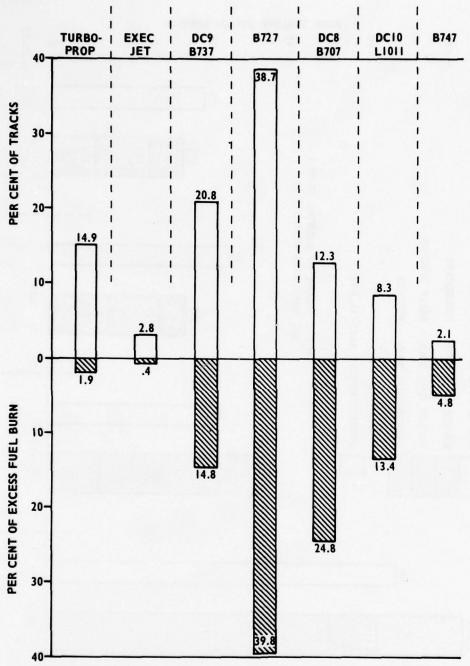
HOLDING SPEED: 210 KIAS OR MINIMUM DRAG AIRSPEED - CLEAN. FUEL FLOW IS BASED ON HOLDING IN A RACE TRACK PATTERN. REDUCE FUEL FLOW BY 5% IF HOLDING STRAIGHT AND LEVEL.

NOTE: IF HOLDING BELOW 200 KIAS IS REQUIRED, FLAPS POSITION 1 AND 190 KIAS MAY BE MAINTAINED WITH A RESULTING FUEL FLOW INCREASE OF 10%.

- 1. SELECT ASSUMED WEIGHT i.e., 90,000 1b.
- 2. SELECT ALTITUDE BAND i.e., 0 15,000 ft.
- 3. USE REGRESSION ANALYSIS TO FIND FUEL FLOW (FF) EQUATIONi.e., FF (1b/hr) = 5112 = 5.125 z + .006  $z^2$ WHERE  $z = \frac{ALT}{100}$  78-40-26

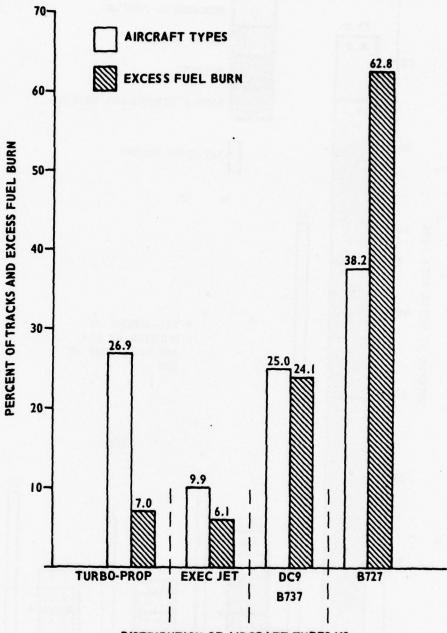
FIGURE 17. METHOD FOR DERIVING HOLDING FUEL FLOW RATE





DISTRIBUTION OF AIRCRAFT TYPES VS PROPORTION OF EXCESS FUEL BURNED BY TYPE (ALL SAMPLES)
78-40-28

FIGURE 19. DISTRIBUTION OF AIRCRAFT TYPES VS. PROPORTION OF EXCESS FUEL BURNED BY TYPE (ALL SAMPLES)



DISTRIBUTION OF AIRCRAFT TYPES VS
PROPORTION OF EXCESS FUEL BURNED BY TYPE (DCA SAMPLES)
78-40-29

FIGURE 20. DISTRIBUTION OF AIRCRAFT TYPES VS. PROPORTION OF EXCESS FUEL BURNED BY TYPE (DCA SAMPLES)

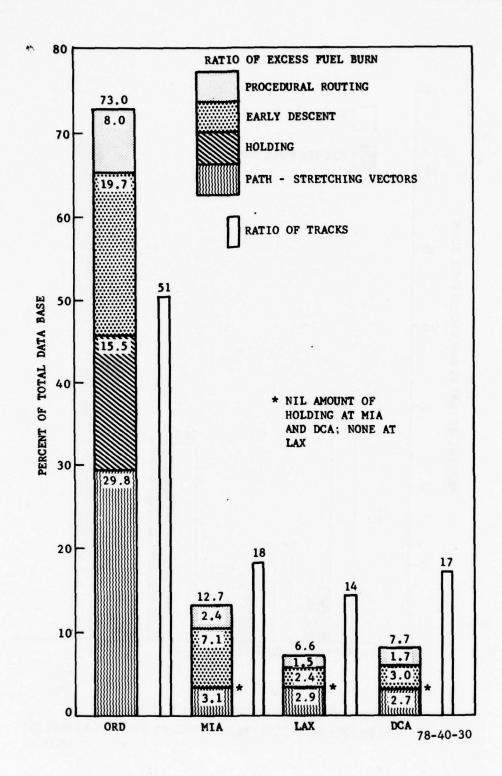


FIGURE 21. RATIO OF EXCESS FUEL CONSUMPTION AND NUMBER OF TRACKS FOR EACH AIRPORT WITH RESPECT TO THE TOTAL DATA BASE

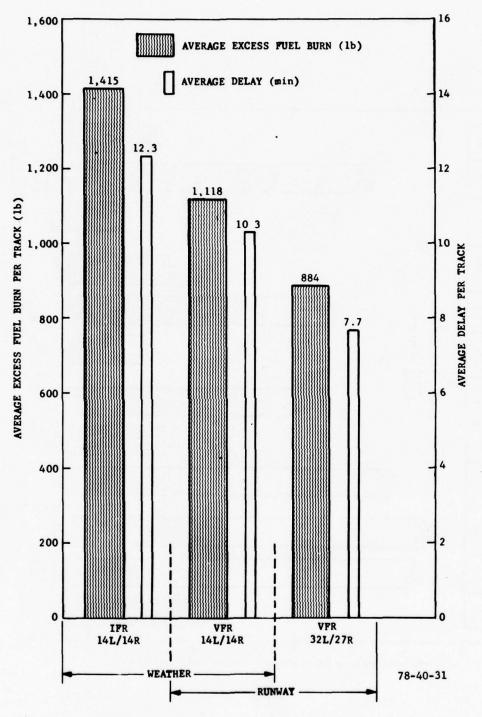
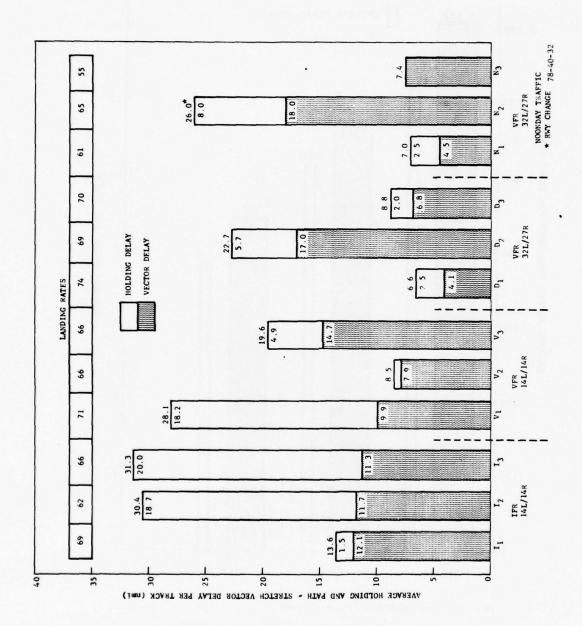
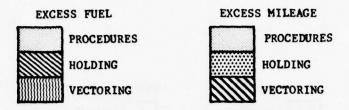


FIGURE 22. EFFECT OF WEATHER AND RUNWAY CONFIGURATION ON DELAY AND EXCESS FUEL CONSUMPTION (ORD)



SAMPLE-TO-SAMPLE VARIATION IN HOLDING AND PATH-STRETCHING VECTOR DELAY (ORD) FIGURE 23.



\* DATA FOR EARLY DESCENT/SPEED CONTROL NOT INCLUDED

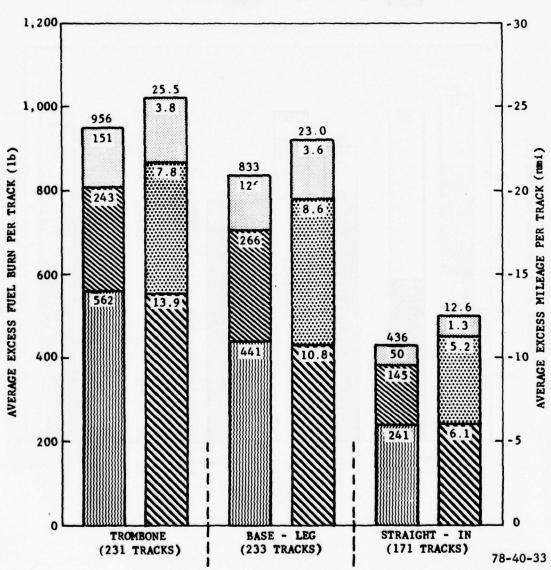
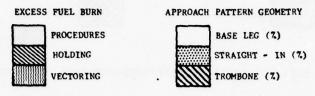


FIGURE 24. EFFECT OF APPROACH PATTERN GEOMETRY ON EXCESS MILEAGE AND EXCESS FUEL CONSUMPTION (ORD)



\* DATA FOR EARLY DESCENT/SPEED CONTROL NOT INCLUDED

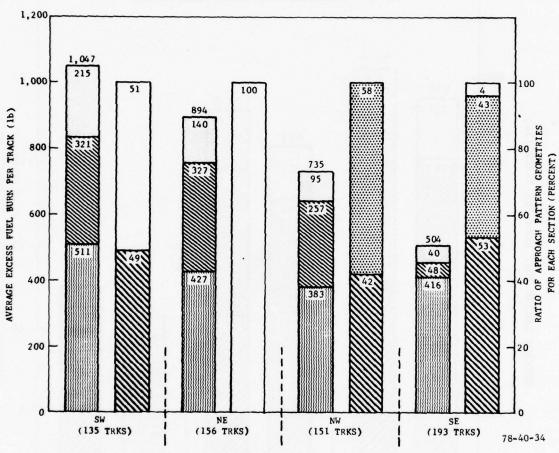


FIGURE 25. EFFECT OF ENTRY SECTOR ON EXCESS FUEL CONSUMPTION (ORD)

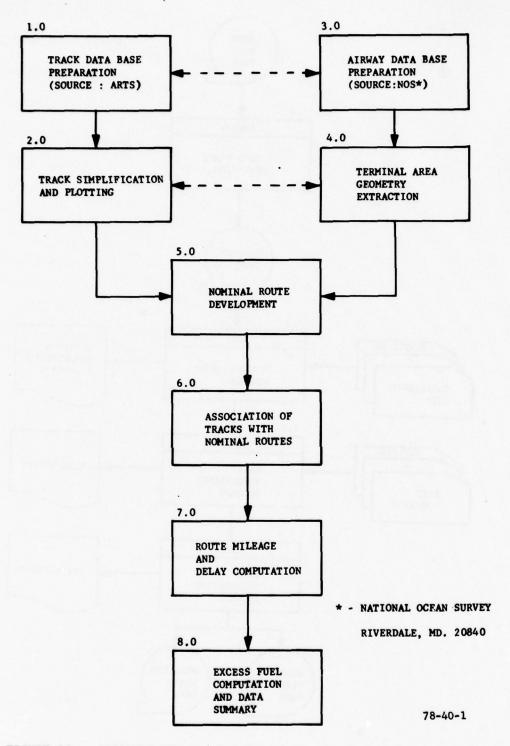


FIGURE 26. METHODOLOGY TO COMPUTE DELAY AND EXCESS FUEL CONSUMPTION IN THE TERMINAL AREA

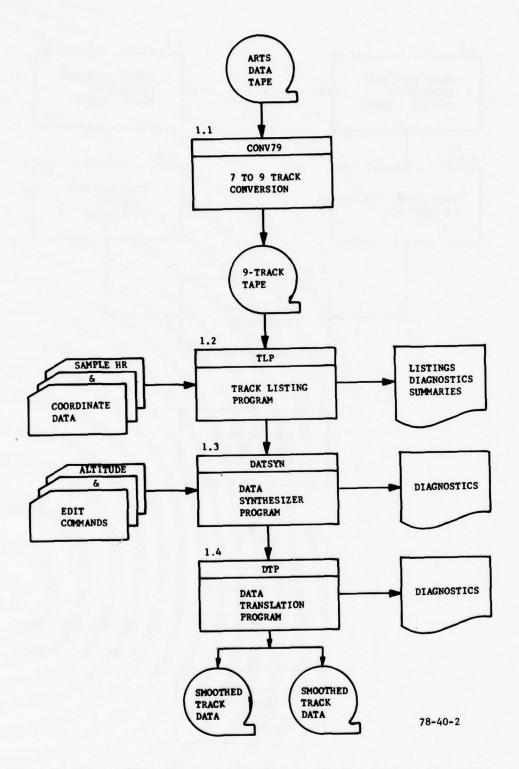


FIGURE 27. TRACK DATA BASE PREPARATION

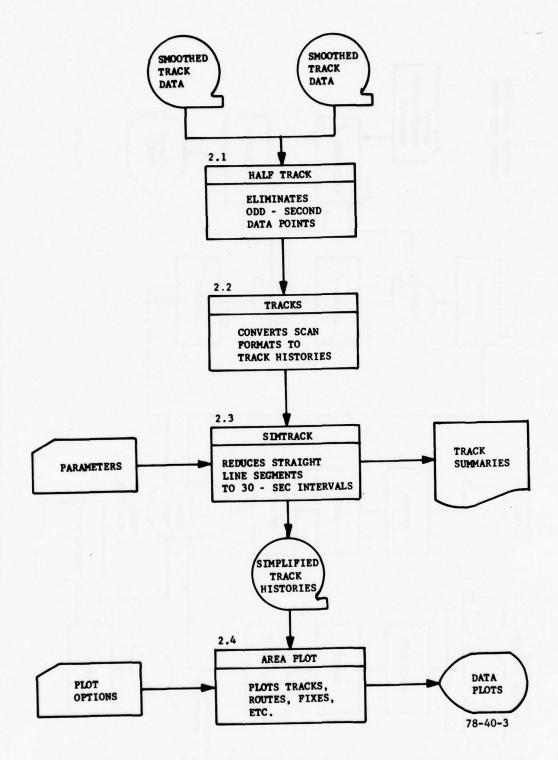


FIGURE 28. TRACK SIMPLIFICATION AND PLOTTING

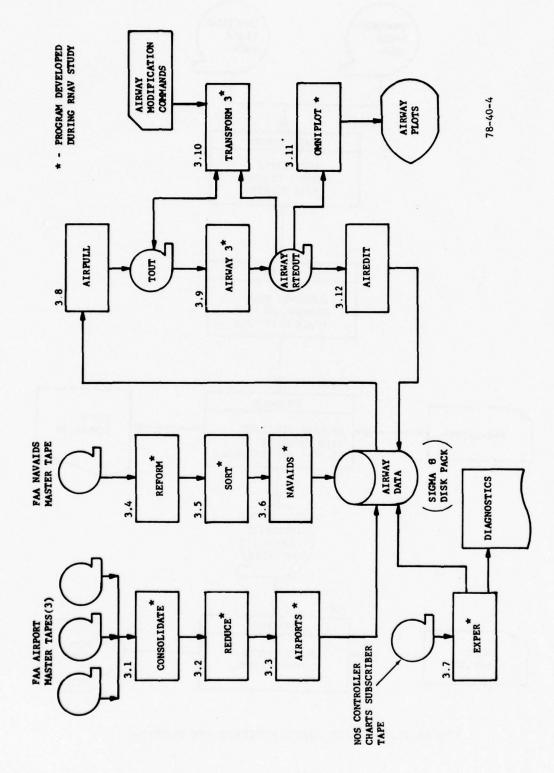


FIGURE 29. AIRWAY DATA BASE PREPARATION

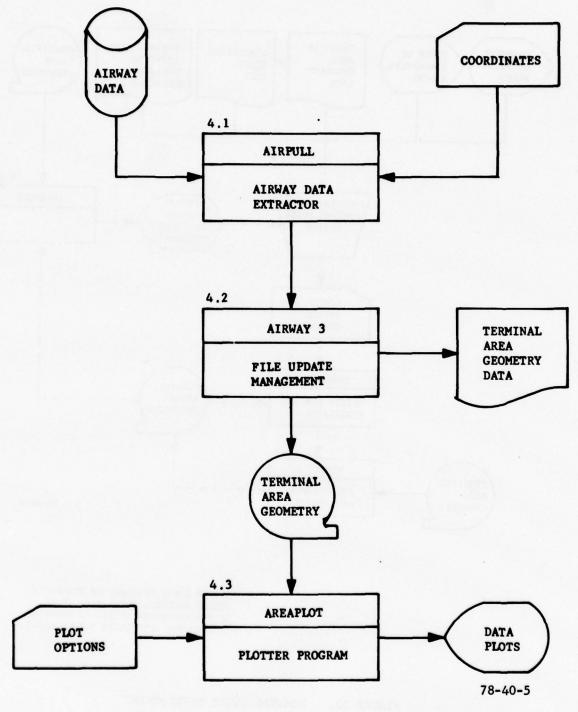
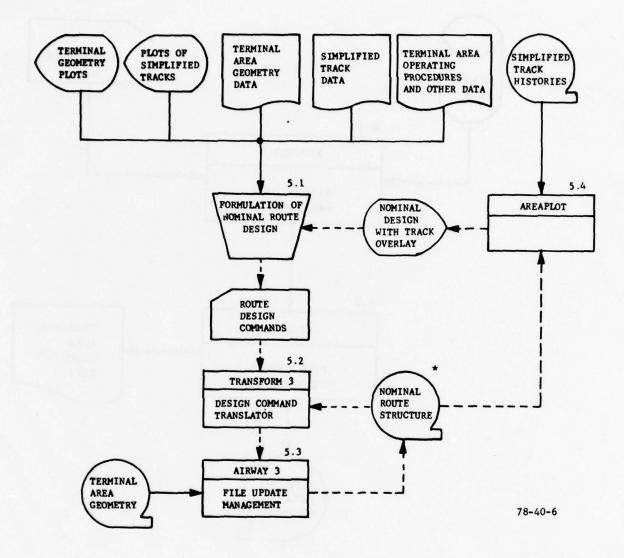


FIGURE 30. TERMINAL AREA GEOMETRY EXTRACTION



\*DASHED LINES INDICATE AN ITERATIVE DESIGN PROCESS, WHERE EACH LEVEL OF NOMINAL ROUTE DESIGN IS MODIFIED TO FORM THE SUCCESSIVE DESIGN LEVEL.

FIGURE 31. NOMINAL ROUTE DEVELOPMENT

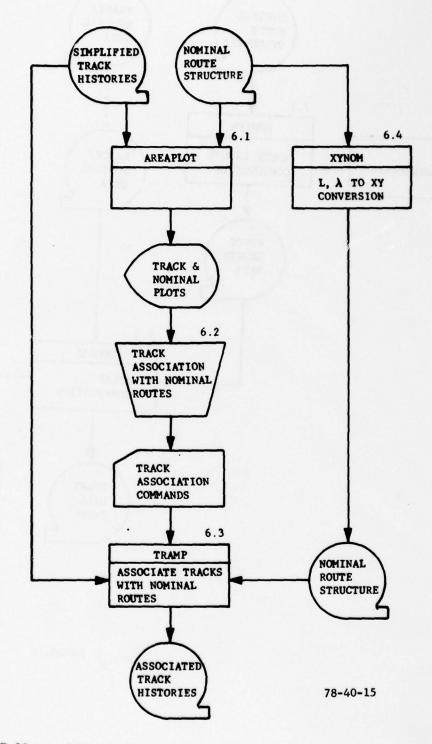


FIGURE 32. ASSOCIATION OF TRACKS WITH NOMINAL ROUTES

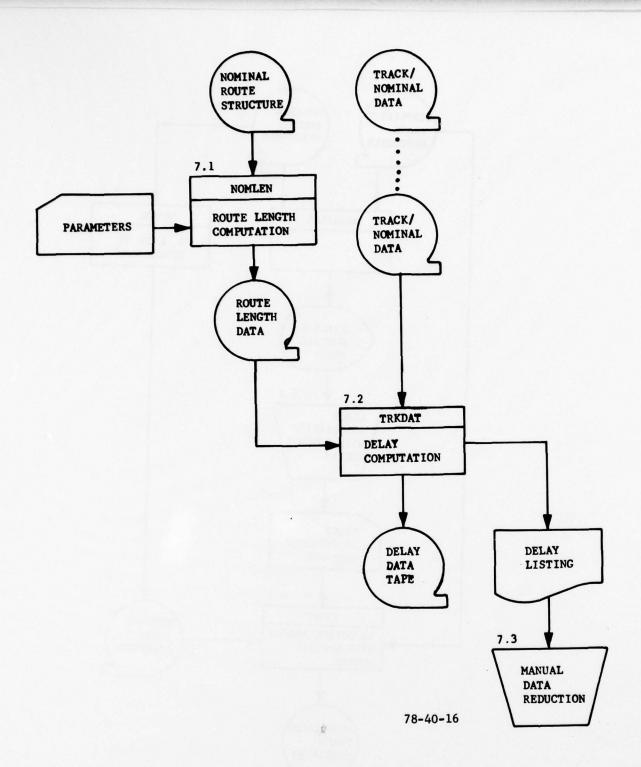


FIGURE 33. ROUTE MILEAGE AND DELAY COMPUTATION

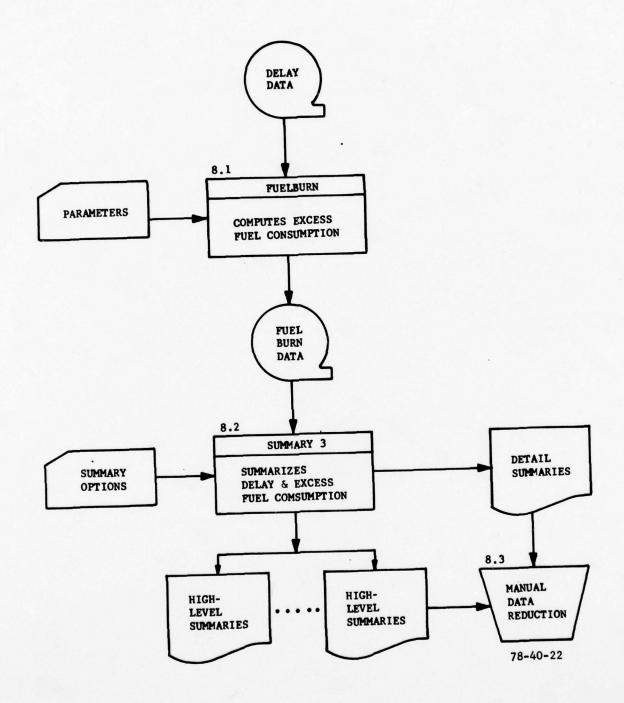


FIGURE 34. EXCESS FUEL COMPUTATION AND DATA SUMMARY

## APPENDIX

Fuel Burn Summary 12 ORD Sample Hours

FUELBUR, SUBSTRIEF	PROJECT: 011=001=230  AUTHERS: RICHARD W. SOPER, ANA-220, THORAS R. CHOTCE, ANA-220	MATIONAL MYLANING FACILITIES EXPRETENTAL CENTER ATLANIIC CITY, NEW JERSEY, 08405	AREA- GRO AREA- GRO

	08103 SEP 16,'77hPMS 15145 WLL 18,'77 TRAMP 20135 AUG 26,177	THACKS IFIAU APH 225'77KFFS 22136 MEP 215'77 TRAMP 21154 MOV 105'77	TRACKS 17:52 APR 26.177APMS 22:48 ALG 26.177 TRAMP 23:25 SEP 02.177	TRACKS ISIOO BEF 07.77KFF6 IIISS NEW 165'77 TRAFF 15129 NOV 223'77	-RACKS 10:29 SEP 20,177hr'S 11:53 NEV 16,177 TRAMP 18:08 NBV 28,177	THECKS ESISS APH 28, 77kPPS 11155 NEV 16777 TRAFF 17:44 NOV 28777	TRACKS 21:20 SEP 08.177hrs 22:36-86F 21.177 TRAP 20:53 GCT 20:177	PFRUED 23:53 BEP 223'/7KPPS 15145 JLL 157'// TRRMP 21165 AUG 265'/7	17154 SEP 261177hers 15145 UL 151177 TRAMP RETZO AUG 261177	PFR0ED 18:26 8EP 273'77KPHS 22:48 ALG 263'77 TRAMP 13:58 8EP 013-177
A BASE GUYP, Y YAPE! . FULLFLEM DATE 22148 PAR 36, 178	7 7 8 9 6 0				> 6				PFRGED	
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THE FELLOWING FILES E	Z. CETAILEE SAMPLE DATA FILE NE. CPINI 1. TAKDAT DATE &C.C. NO. (THIS CEPIED FILE NAE CETAILEE	3. DETAILED SAMPLE CHIAI 2. TRKDA!	4. CETAILED SAMPLE CF1s1 3s TRKDAT ITHIS COPIED FI	C. CETAILED SAMPLE CHAV 1. TRKDAT (THIS CEPIED FI	6. CETAILED SAMPLE DATA PILE NO. CHIAV 2. TREDAT DATE 22:11 NO. (THIS COPIED FILE NAS DETAILET	7. CETAILED SAMPLE CHINY 3. TRKDAY ITMIS COPIED FI	S. DETAILED BAMPLE CPISO 14 TRKOA (THIS COPIED F)	S. CETAILED SAMPLE CHIEG Z. THEORIE ITMIS COPIED FI	16. CETAILED SAMPLE CHIO 3. THROAT THIS COPTED FI	11. CETAILED BAMPLE CPION 1. TRKDA

ITHIS CEPIEC FILE MAS DETAILED DATA INPLT FILE NG. 10)
AR. ESTAILED SAMPLE DATA FILE NO. 11. COPIED FILE: CPINA 2. TARDAT DATE 22750 DEC CLAPTA PRAPTICALA 2 ANDED 18153 BEF 27/1976/PB 2218F ALG 28/197 TAMP 13125 BEP DIOTYP
13. CETAILEC SAPPLE DATA FILE NE. 12. COPIED FILE: CHIN 3. TRKDAT DATE 22127 CEC 01.77, TRAPSIGNIN 3 "FRGED 09:92 SEP 28.177hry5 22:08 ALG 26.77 TRAPP 21:03 AUG 30.177
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		Ev 7.408 497	7.425	.40.	7 38	26.030	268	21.335	4:25	200	10.468	8818	
TETAL   492-363 21375 414-442 18412 77-921 2942 1456-812 49908 997-884 244104 29117 864-86 20180 8378   ACTIVALBRDVOZAV- PAIRBECARK- Rhw14L- TVPET	TETAL   492-363   21375   414-442   18412   77-921   2942   1456-512   49508   957-884   244104   29117   864-86   20150   8378	E 11-190 485	9.419	18 1.77	. 67	33.,85	1134	63.888	14:56	1941	17.297	*110	
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### 1 ### ############################	TRK-DVERGERIN TRK-CVER-CACH CACH-OFR-CFIN PATH-STRETCH TYPE 1 MAYC HOLDING 1 177E LBS 1 AT 12 LBS 1 AT	NAL ORDVOZAV. PAIR ORC	ARE APEY.	.A14L, TYPE	**************************************	IN 74.735.	ADURON	77.442,	INCR 2.	706 3.6			
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0001 1943 40141 69522 45234 1943 6952 5503 1943 695244 1943 695244 19434 1966 69524 1966 69524 1966 695244 1966 695244 1966 69	0000 1847 65524 85652 55113 95150 CAFT 05155 5555 4515 000 000 000 000 000 000 000 000 000	18:191 770	8824/3	38.	-	18.18	508	182.20	17102	-	0000	-	
		17 672-358 28520 61	162 200-66	35.00	E.E. 0	682.639	16662	186.24	20161	11/2	0004		

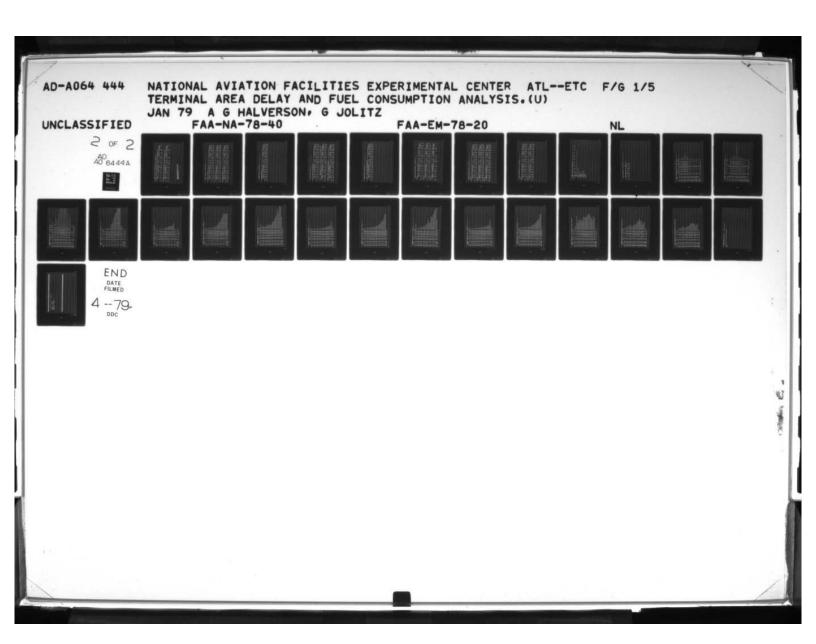
	NETILAL. CROCOLOVE PAIR. CENERRE, Rhut. AETR, TYPE. B. ACJESM 55.454, ACJESM 64.271, INCR 8.877 16-CR	PAIR	CACARR	RANT	7R, TYPE.	8, ACJR	IN 55.49	ADANG!	64.371	INCR	1.877 16.	23		
	TRK-CVER-CMIN	CHIN .	C.3 T CF CAC A	F.CNC*	CHIN THRECKER CASTONED TRACKS CHIN THRECKER CASTONER CHIN	· CPIA	PATE STRETCH	456	TYPE	A HAVE HEL	A HAVE HELDING ;	Jahr 2	TYPE AS E TYPE TYPE '8' HOLDING NP	
\$4.0EV	982.6		834.6	-	1.042	210	11.929	228	13.678	31116	1881	12.168	9212	163
	18-170	178	10.366	181	1.804	6	12.946	446	35.630	9145	1981	595.62		*
1 1	TETAL 1162-503	186+	463.425 31019 CREARS ONLY	11019	36431 844-664	16,31	743.57	36934		15:30 38:00	1016	080-86	HEIEL	
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=	304.217	11083	28C-11C	10228	24.107	500	34744	11682	25.754	9:0	3	41.580	10:00	3
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THANY BY NOPINAL CP   PE TRACKS AND	######################################			NEP INAL O		STOEEV	319	16741	NEP INAL BR	- 0F	1.66,	- AVE	TETAL 15	OF INAL		BYACEV	346	TETAL				

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ACH NB V 2. 2. AC UH I		356	20332	ABUMI	FIN LBS -	- 62	- 75	1415	ADUMI	411	1 297	E	£ .	1864	EST 0	USHE	
PPARY BY NOWINAL OF THE TRACKS ASSIGNED TO EACH NOFITHAL	THE CYER CASTONED TRACKS THE CYER CAST OF CYER CFIN	258.	450.320 2	RECOUPEL, PAIR "CELIAR, ARKY AJZE, TYPE B. ADUMIR 66.727, ADINCY 61.461, IRCR 1.734 2.64	# C % CF CRC ASSIGNEC TRACKS THK CVER CLOFF CNOFF CPIN NF LBS   NF LBS	.716	.788	40-172	RDOOBAV, PAIRCRUARS, RNHVA14R, TYPE8, ADUMIN 55.338, ADUMGH' 60-568, INCR 5-230 9-53	TAK-EVER-CASTIGNED TRACKS	-	5.00.2	3.591	79-152	HHIS PAGE IS REST QUALITY FRACTICABLE	MINON COPY FURNILSHED TO DDG	
ACKS AS 1, 1 2	CNOP	525	36640	hy A32	CABE	+30	512	26126	¥	RC ASSI	 Se 1	+16	5/2	1919	THIS	FROM (	
27 17 18 18 28 28 28 28 28 28 28 28 28 28 28 28 28	THE CERCENT AND	15.637	E57.128	CECARE, AN	THK-EVER-	8.5.8	13.098	867.578	CRLARE, RN	THE CVER-CARP	-	162.6	116.1	136.802			
20 0 11 10 4 10 10 10 10 10 10 10 10 10 10 10 10 10 1	CHIN LBS 1	999	56972	PAIR		430	539	27538	PAIR		 -	533	478	10535			
CHTERY BY	TRESTRACES ST	22.938	1307-448	. GRC0078V	F TRACKS 51 TRK-OVER-CHIN	8.337	13.885	708-150	. ORDOOBAV.	CKS VER,	Ę	9.125	9.912	218-057			
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ETICE'S 5-56-3 1176 5-877 1080 1-177 161 12-544 1070 5-182 1135 439 -0000  AVE 14-517 756 5-857 1080 1-177 161 12-544 1070 5-182 1135 439 -0000  TOTAL 372-918 18972 2-35-101 13822 133-817 5149 347-748 16727 108-687 2-9104 8904 -0000  TOTAL 372-918 18972 2-35-101 13822 133-817 5149 347-748 16727 108-687 2-9104 8904 -0000  TOTAL 372-918 18972 2-35-101 13822 133-817 5149 347-748 16727 108-687 2-9104 8904 -0000  TOTAL 372-918 18972 2-35-101 13822 133-817 5149 347-748 16727 108-687 2-9104 8904 -0000  TOTAL 372-918 18972 2-35-101 13822 133-817 5149 347-748 16727 108-687 2-9104 8904 -0000  TOTAL 372-918 18972 2-35-101 13822 133-817 5149 347-748 16727 108-687 2-9104 8904 -0000  TOTAL 372-918 18972 2-35-101 13822 133-817 5149 347-748 16727 108-687 2-9104 8904 -0000  TOTAL 372-918 18972 2-35-101 13822 133-817 5149 347-748 16727 108-687 2-9104 8904 -0000  TOTAL 372-918 18972 2-35-101 13822 133-817 5149 347-748 16727 107-84 5-101 17-84 5-10	1.177 16 5.353 20 133.817 514 68, TYPE:-5, AC	161 16	13.912   10 13.912   61 13.912   61 13.912   62 14.614.61467	1070 1070 1070 1070 1070 1070 1070 1070	201172 271172 106:687 34:737.1	# 1196   198	000 000 000 000 000 000 000 000 000 00	**************************************	
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TRY CF CFC +88  VER CF 11 TRY CF CFC +88  LBS   TRY CF CFC +88  LBS   TRY CF CFC +88  CBS   TRY CF CFC +88  CBS   TRY CF CFC +88	CAGE TRACKS CAGE OVER CP		ATH. STRE		TYPE 14	HAVE HELDING HELDING TIPE LES	C TYPE	AAG 24 TVE	
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	.330	50	18.401	732	0000	-	13.259		
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424.264 20899 358.737 19726	25.527 11	1173 116	1126.494 4	41500	0000	-	727.957	248134 2	21774
NEPINAL GREDOSBVA PAIR GELARFA GNAV ARZEZ TYPE TYPE TA AGUTIN 72,393, ADUNEP 751332 INCR 21779 3:85	7R, TYPET.	A1 45	72,353,	A02ha7	75.133, 1	INCR 2-779	3.68		
OF FRACKS DB 6.C A OF CRO ABSTONED VERTERS TRK-CVER-CNOT CNOT-GREEN	TONED TRACKS CNOW-SVER-CFIN		PATH. STRETCH	3	01 3418	10 HAVE HELDING :		TYPE AS 2 TYPE	
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**4** □



*** SUPPRAY BY NOTINAL EFTEE TRACKS ASSIGNED TO EACH NOMINAL *** SAPPLE : OND 12 COMPENIS : CHI SAPPLES : 1 1, 1 2 2 1 3, V 1, V 2, V 3, C 2, C 3, N 1, N 2, N 3, NETNAL CORPOLONY PARR CEREARS ANNY ALL, TYPE. S. ADUPIN 55,706, ADURE 60 8619 1KR 5.185 9-38	THE TEST O TYPE AS O TYPE BY THE LIST OF T	999•	0000 821	REFINELGROGIOBY, PAIRERUARS, RALVAZYR, YFFEB, ACJFIN 63.778, ADJREF 64.093, INCR .313 .EX	O HAVE HELDING : C TYPE AJE O TYPE B	LOS I APE '8' HOLDING	0 000 0	0000	0 000.	·	
oss ELPPARY BY HOMINAL EF THE TRACKS ASSIGNED TO EACH NORTHAL see SABTLE: OND 18 CHPENTS: CHPENTS: CHPENTS: CHPENTS: CAPPARE: OND 18 N. 9.	TYPE 'A' PELDING	000.	9616 998-12 9616 998-12	64.033 INCH	0	APE 'A' EELETKS	000•	000.	000•		
C 2 C	ETCH LBS	- 80	2 6	ancor.		LBS 1	72	- 2	1177		
MINAL	PATH, STRETCH	62.06	52.197	N 63.778		PATH.STREYCH	6.485	8.412	129.186	PE	
EACH NO	CPIN	-	F 6			CP IN	-		2	ICABI	
S. 13, V 1 L, TYPE.S	1.7 % CF CAC ASSIGNED TRACKS THE CVERSCACY CACY-CVERSCACY NAT LBS	1.883	1.908	R, TYPE.	2.4 % CF CRC ASSIGNED TRACKS	Cherever-CPIN	.139	-115	1.729	THIS PAGE IS BEST QUALITY PRACTICABLE	
ACKS A	CACE LBS !	- 2	198	244.	SRC ASS	LBS	78	- 82	1177	MALE	
CALARA R	THE CYER CACH	5.685	36.329	CHUNKS R	** * **	THROCVER	6.588	B.612	129.186	S BEST	FIGURE COLY PURPLEMENT OF THE PARTY OF THE P
PAIR.	CFIT LBS	- 1:	359	PAIN	1.5	LBS 1	**	2	1200	PAGE 1	1
CGHERTS :	F OF TRACKS 11 TRK-EVER-CFIN	6.244	*:665	GREGIOBV	8 OF TRACKS 15	TRK - CVER -	609.9	8.728	130.915	THIS	MORN
CFINAL.	-	STOCEN	1611	CP THAL.	•	-	STOCEV	- 3vE	TETAL		

SAPLE: GRO 12 22.13. V 1. V 2. V 3. L 1. D 2. D 3. N 1. N 2. N 3.  22.13. V 1. V 2. V 3. L 1. D 2. D 3. N 1. N 2. N 3.  310AED TRACKS  NH LBS   NH	### FOR THE AIRCRT CRC  ### CASIGNED TRACKS    TRACKS 233
ESTRPE : GRO 12 EST 1 32 V 13 V 23 V 34 F 13 E 24 CAGH-OVER-CPIN PATH-STRETCH NH LBS   NH LBS   3-77   175   19-8.37   78- 3-562   126   19-8.37   78- 3-562   126   19-8.37   78- 3-562   126   19-8.37   78- 10-80   18-80   18-73   18- 10-80   18-80   18- 10-80   18-	THE AIRFERT CRC  36-73
E. 1 3 v 1 v 2; E. 1 2 v 1 v 2	HE ALFERT CRC.  36-7 X CF CRC.  36-7 X CF CRC.  10-81 3-92 X CF CRC.  10-81 3-92 X CF CRC.  10-81 3-92 X CF CRC.  28-5 X CF CRC.  10-81 3-92 X CF CRC.  10-81 4-91 3-92 X CF CRC.  10-92 X CF CRC.  10-93 CF CRC.  10-94-81 4-12-8 29-96
	10-51 CRC AST

4405 156 246 2 CF CRE A STOKE TAACGS  4405 156 246 2 CF CRE A STOKE TAACGS  4405 156 246 25	FRICKS 156
	CKS 156  CR CATA  CR

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444 SLHPARY BY SECTOR 44* SAPPLE ? GRD 12 COMPENTS : CHI SAPPLES : 1 1 1 2 1 3 7 4 3 7 5 7 3 7 5 7 5 7 5 7 1 1 N 2 2 N 3 3 7 5 7 5 7 5 7 5 7 5 7 5 7 5 7 5 7 5	179E 'A' PELSIAC TYPE ALE 29 TYPE  179E 'A' PELSIAC TYPE 'B' HOLDING  177E LW		
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DV SECTOR ****  NIS : CHI BAPPLE  TO CHI	6 OF TRACKS 151 23-8 1 CF CRC A8810 TRK-OVER.CHI. THK CVER.CNOP C. LOS I N. LOS I N. COS I C. 145 657 ST-DEV 10-431 703 1C-147 657 AVE 11-037 478 8-634 383 TETAL 1666-604 722-6 1303-766 57878		
SCHPEN SCHPEN SECTOR	6 OF TRK. 10 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1		

		0199V 3	7 CF ORC ASSIGNED TRACKS	5	S 30 4-7 2 CF CRC ASSIGNED TRACKS S 30 4-7 2 CF CRC ASSIGNED TRACKS S 30 4-7 2 CF CRC ASSIGNED TRACKS			3 PAVE PELDING	E SILO	1 Type	A.6 & TYPE	
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1130   21   21   21   21   21   21   21   2		- 82	2.38.2	-	8.882	- 2	0000		- 0	1.502	01+2	11
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16136 899 16136 899 16136 899 178E 600	99	- 96	3.406	*	110.01	78	9.306	9:03	- 00	7.261	1:99	•
### ### ##############################	85	- y	3-161	- 22	1026	-6,	24.984	6:20	171	12.521	3139	7.
11 PEC D 1 PE D 1 PEC	1 6292	+ 590	240-271	1702	792.188	5574	99.936	33:36	•	60.08	1+136	888
1 THACKS 5 .8 3 CF CRE ASSIGNED TRACKS  THACKS 5 .8 3 CF CRE ASSIGNED TRACKS  1 THACKS 1 THAC	1											
TALE NOT THE CHEST CHEST CHEST   TALE   TALE   LBS   TA	. 5	1C A881G	NED TRACK					DH NEVE HO		7		
3.799 21 3.783 21 .589 5 3.783 21 .000 C .000 5.128 29 3.892 22 .000 0 .000 25.428 145 19.459 11C 6.182 35 19.459 11C .000	R.CHIN LBS -	רפו ו	NOT SUCK.	1 987	NP STR	1 587	34	TATE TATE	- 81			3
5-128 29 3-892 22 1-236 7 3-892 22 -000 0 -000 25-6+1 145 19-489 11C 6-182 35 19-559 11C -000 C -000	512	12	. 989	•	3.783	12	000.		- 0	0000	-	•
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240 1C-5E1 277 2-583 62 19-751 441 48-419 11121 882 35-656 8121  24855 1445-456 36451 340-998 8213 2607-145 58228 48-159 153130 8884 677-456 228134 12  222 36-52 255 255 255 255 255 255 255 255 255	240 10.551 277 2.583 62 13.751 041 48.419 11121 882 8 44895 1440.456 36461 340.958 8213 2407.145 98228 484.159 155130 8824 67 222 34.5 18.5 18.5 18.5 18.5 18.5 18.5 18.5 18	13-63-4 1764-494 44 1764-494 13 19-292 19-29
######################################	### 125   1446-456   34661   340-558   8213   2407-145   58228   484-159   153130   8824   67    ###################################	1786-494 444 444 444 444 444 444 444 444 444
229 36.5 % CF DR. ABBIGNED TRACKS  CHIN TRACKER-CRUP CNURT-CPIN PATE, STRETCH TYPE 'A HERCITA 1965 CAS 22 TYPE LAS 17.21 4101  S29 36.5 % CF DR. ABBIGNED TRACKS  248 3.342 32.5 418 3.229 316 33.553 633 27.014 6:36 763 37.121 4101  S26 3C % C % OF CRC ABGIGNED TRACKS  258 36.C % OF CRC ABGIGNED TRACKS  S26 3 250.C % OF CRC ABGIGNED TRACKS  S26 3 250.C % OF CRC ABGIGNED TRACKS  S26 3 250.C % OF CRC ABGIGNED TRACKS  S27 3 418 3.23 210 22.0 % 1047 24.806 3129 994 19.286 4180  S28 3 30.C % OF CRC ABGIGNED TRACKS  S28 3 3 3 3 3 3 3 3 4 3 4 3 4 4 4 4 4 4 4	232 34.6 % CF CRE ASSIGNED TRACKS  CHIN	13 10 10 10 10 10 10 10 10 10 10 10 10 10
232 30.6 % CF CRC ABSIGNED TRACKS  CHIN TRIEVER-CRUP CNUR-CVERTCPIN PATE,STRETCH TYPE 'A HELDING : 22 TYPE 2.6 32 27.014  279 9.125 36.2 % CR CRUP CNUR-CVERTCPIN PATE,STRETCH TYPE 'A HELDING : 27.014 6.106 763 17.121 4.101  835 1C.553 416 3.259 116 13.55 63 40.074 1138 1487 30.446 7105  848.3 25.014 6.104 36.03 30.446 7.05 36.446 7.05 36.446 7.05  848.3 25.014 6.104 36.01 30.446 7.05 36.446 7.05  848.3 25.014 6.104 36.01 30.446 7.05  848.3 25.014 6.104 36.01 30.446 7.05  848.3 27.014 6.104 30.446 7.05  848.3 27.014 6.104 30.446 7.05  849.3 20.446 7.05  849.3 20.446 7.05  849.3 20.446 7.05  849.3 20.446 7.05  849.3 20.446 7.05  840.3	232 36.6 % CF CRE ABSIGNED TRACKS  LSS	10-12-4 10-12-
### THE TABLETON FORMACTION FOR THE TATESTING TO THE THE THE LAST HALLING FOR THE LAST HALLIN	CHAIN THREVERSCROP CHAPTERIN PATH, STRETCH IT THE LIBS IN THE CHAPTERING CHAP	10-184 10-184 10-184 10-184 10-184 10-184 10-184 10-184
279 3-125 362 3-392 116 15-751 631 40-074 1138 1487 30-445 7103 28-26 15-253 28-614 88 17-121 4101 88-26 25-35 28-573 158138 1057-625 416:04 32037 374-239 346148 28 28-26 28-26 28-273 158138 1057-625 416:04 32037 374-239 346148 28 28-26 28-26 28-273 158138 1057-625 416:04 32037 374-239 346148 28 28-26 28-279	15 379 3-125 362 3-342 121 20-123 633 27-014 6:04 763 17-15 52 54 52 57-014 6:04 763 17-15 52 54 52 57-15 53 54 52 54 52 57-15 53 54 52 54	9-205 1292 1
\$256 16-553 416 3-259 116 15-51 681 48-074 11138 1487 30:448 7105 84263 25264 2616104 35007 374-239 346148 28	12 536 14.553 416 3.259 116 19.951 681 48.074 11138 1487 30.4  66 124263 2554.458 97179 765.258 27084 4582.973 158198 1057.625 416104 38037 974.2  56 124263 2554.458 5818AED TRACKS  CKS 95 156.7 OF CRC A8818AED TRACKS  CKS 95 156.7 O	14-292 3316-766 1204 284 5 15 184-0-0-0-0-0-0-0-0-0-0-0-0-0-0-0-0-0-0-0
18 18 18 18 18 18 18 18 18 18 18 18 18 1	16 184263 285C-4C6 37179 765-258 27084 4588-973 158198 1057-625 416104 38097 974-2 5 CK6 95 35-C 3 OF CRC 48819AED TRACKS CK6 95 35-C 3 OF CRC 48819AED TRACKS TOWN CHILL THE TRACKS TOWN CHILL THE CHILL	3316-766 1244 16 178 6 18 19 19 19 19 19 19 19 19 19 19 19 19 19
4180 4180 4180 6132 1	CKS 95 18-C 3 OF CRC ASSIGNED TRACKS  VENCEPIN TWATEVERSTAND TRACKS  LDS N TTHE LBS N N TTH	OF TRACKS STANKED VERICES
418 TYPE 11ME 11ME 11ME 11ME 11ME 11ME 11ME 11	THE SS 18-C 1 OF CRC ASSIGNED TRACKS PAIN, STRECH 17F A BUCDING 1 4 TY LOSS I NF 17F LOSS I NR L	OF TRACKS 9 THK-DVEH-CH NH LI
16-36- 108- 18-38-7-4 668 3-331 210 22-5-0 10+7 2+806 3125 894 19-886 4180 16-36- 108- 18-36- 108- 18-36- 1	SULLY TREATMENT TO THE TOTAL TRANSPORT TO THE	-
10-124 731 3-544 648 3-331 210 22-540 1047 24-806 3125 894 19-886 4180 16-364 1054 13-548 13-354 13-		- :
18-876 848 3-387 806 21-483 1193 73-482 19111 3191 28-037 6132 1888-749 80632 321-791 19592 8031-348 113347 293-929 116144 12766 BDN-670 187144	731 3-5-4 000 3-331 KIO KK-0-0 1047 K+-000 31K9 884	
1828-749 86632 321-791 19592 2631-346 113347 293-329 116;14 12766 B04-670 187144	1084 12.976 848 3.387 206 21.483 1193 73.482 19111 3191	_
	1828.749 SC632 321.791 19592 2031.348 113347 293.929 116:44 12706	1554-640 1001

- 200	ER-CFIN		CRC ASS	7 R CF CRC ABRIGNED TRACKS	•s*				14 PAVE HELDING	. 941013	3 1796	11 T	
-			- Ch 67	Char. BVER	- CP IN	PATE STRETCH	100	3444	I TOPE	- 887	APE 'S' HOLDING	TINE	100
·CEV 9.17	029	9.035	929	2.890	109	19.963	1004	22.785	5:20	1090	16.497	11811	369
ave 11.76	791	9.243	629	2.525	162	17.408	1001	64.522	15:10	3008	23.228	8118	1000
171 647.22	4 43543	£c4.338	34599	138.886	8943	957.,16	58088	193.567	+813c	9006	289-511	97:46	1100
CATEGERY # 19		61.0											
. OF TRAC	100	1.6 2 CF	CRC ASS	IGNED TRAC	× ×				2 PAVE I	ELCING :	1 TYPE ALE	7 1 34	. 344
- AR - C	ER-CHIN	TAKOROCKET CNOTORENOC	LBS	CNOP. CVER.CPIN	- 661	PATE STRETCH	- 887	JAA.	TYPE 'A' HELDING	- 100	-	TYPE 'B' HOLDING	NG P
.DEV 11.27	0 1616	11.909	1633	2.187	307	14.913	1625	0000		ر ا	000.		0
4VE   22.06	0 2686	15.803	1941	6.257	74.	22.194	2482	+0.301	10:00	3876	26.605	6130	2139
ETAL 220.60	2 26860	153.631	19418	62.571	75.5	224.937	24828	+0.301	10:00	3270	26.608	0130	2139
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3/4 3/4										
3/4				,					.	
2, 7 3, 4/5 4/9E 4	91 74									
1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	2 63									
2 2 3	-									
1, 0 2, 4/C	- 2 4 8									
000 COUNTS BY AIRCRAPT TYPE 000 SAMPLE : OND 32 COMMENTS : CHI SAPPLE : 13, 12, 13, 14, 17, 17, 17, 17, 17, 17, 17, 17, 17, 17	~									
V 12 V 18	-									
A/C + 4/C	0159 2									
8 1 2 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	FA27 15									
4/C - 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1		-		-					*	
RCRAPT TY CHI SAPE A/C TYPE 8	- 0 - 1 - 2 - 2 - 2 - 2 - 2 - 2 - 2 - 2 - 2	96 630	900	21,101,						
CONTRES 2 42 CONTRES 2 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	CVSB 32 TETAL - 76	8737 36 707AL = 132	8787832 1014L • 232 1 1 1	DC10 43	8747 10 TOTAL - 10					
CATEGERY AVC	- V - B		13 67	19 - 19 A	19 67 A					
					•					

133 ASSIGNED TRACES PARE PETERNO 1 45 TYPE 181  1
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COTTENTS : CHI SATTLES : 13 : 15 : 15 : 15 : 15 : 15 : 15 : 15	CHI SAFTLE	1	1 69 1				2 2	10 4 10 4 10 4 10
ALTITLDE BAND	EN EN	AVERAGE	- 887	-	TOTAL	BE ICEURT	AVC .	
99995 X X 11499	146.69	16129	-66	393-427 138156 11276	138156	11876		*****
118CC 5 x > 10999	29.62	9 : 90	-04/	529.62	9190	-02	-	•
1106C 3 x > 10499	31.515	17:30	1912	816-18	17:30	1916	-	•
108CC > x > 9999	283.94	16164	1136	138-249	30:12	0440	-	
1000C X X 2499	*50.84	11:30	- 111	481.96	23:00	6824	2	••
98CC 3 x > 8999	134.65	10101	1282	988-862	59110	1684	•	
900C 3 x > 8499	49-611	11160	1	+92-122 201100 1820+	201:00	15564	F	
880C 3 x > 7999	48.258	11:52	1642	386.068 135:02	135:02	13136	-	
300C 2 x 2 7499	\$12.at	9149	1861	144.855	39:00	+264		• • •
78CC 2 X 5 6999	48.300	Falls	=	1691219	48134	deck	-	
700C 3 X 5 6439	230.	-	•	999.		0	5	
680C X X 2088	298.12	188	621	198-12	1534	2	-	•
800C 3 X 3 2439	0300	-	-	0000	1	-	5	
560C 2 X 5 4999	2000	-	-	000.	-	-	5	
800C 3 X > 4499	2200	-	-	000.	-	•	5	
480C 3 X 3 3899	330.	-	-	999.	-	-	5-	
400C 5 X 5 3499	2000	-	o -	000.	-	•	-	
380C 5 X 5 2959	200.	-	-	000.	-	-	5	
3000 3 X > 8499	2000	-	-	900.		-	5	
180C 5 x 5 1939	200.	-	6	900.	-	-	0	
200C 3 X > 1499	200.		-	.000		-	5	
1800 5 X 5 00 5	200.	1	-	990.	-	6	-	•
0 4 X 8 5666	262.25	18103	1483	201206 Sterigia	902188	10/00		

1   1   1   1   1   1   1   1   1   1		*** SAMPLE : 6RD 1E V 12 V 22 V 32 N 12 D 22 D 32 N 12 N 22 N 32 TOTAL   A/C   F TIME   185   COUNT	0 0		NO 232 1 6	2 .003 2 6 6	 6 2319 5 6 4 4 4	\$ \$ \$ \$ \$ \$ \$ \$ \$ \$ \$ \$ \$ \$ \$ \$ \$ \$ \$	9 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0	2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2	19767 21 600000000000000000000000000000000000	2 215 2 6 4 4	* * * * 1 * 002* *	9 9 9 9 9 9 9 9 9 9 9 9 9 9 9 9 9 9 9 9	9 9 2 922		9 1 6Gt 91	0 43 11 6		5			18 7ABUS 87
	Tate of the control o	18: *** SAMP 3. V 1. V 2. V TOTAL NP TOTAL	2000	000.	13.609 3100	33.690 7:32	 82.106 20:16	184.412 42:48	158-313 36:36	705-292 245122	687-002 241138	101.746 23:32	11:25 25:14	145-176 33:56	94-339 21:16	200.	96:5 688:22	00:2 526.9	200.	200.	900.	200.	0:506 882.965:0
	19 666.92 19 666.92 19 700. 100. 100. 100. 100. 100. 100. 100.	I 1 1 2 1 I	0-0	- 0		_	 														-		

A   A   A   A   A   A   A   A   A   A

COMPENS:	L L L L L L L L L L L L L L L L L L L		1, 1 2, 1	3, 4	2	COFFERS CHI SAFEES 1 1.1 2.1 3. V 1. V 2. V 3. C 1.0 0. C 3. N 1. N 2. N 3.
PILEAGE BAND	AVERAGE	36	10741	- 1	1 4/6	
		5 -			200	
16 CX C 5665	700.	-	000.	0	0	
91 > x > +8	*8.93*	2372	<b>*8.93</b> *	2372	-	
50 X X C 30	2000	0	2000	0	0	
45 7 X 7 48	48.615	1875	128.457	3827	-	3 1000
46 3 x 3 39	\$39.04	2100	162.018	1642		*
35 > x > 36	37.580	2230	150.319	2261	•	****
36 2 x 5 33	30.369	103	34.369	709	-	•
3: 2 x 5 30	313-116	199	186.061	9644		******* 9
3C 2 X 7 27	201.05	1364	+50-320 20864	19802		***************************************
27 X X 242	164.62	1265	200-803 27848	202/2		
RANN EL	216.34	68	29662 9/6-186	29662		32   oceances acts and participation of the contract of the co
21 5 x 5 18	18.48	195	837.869 34202	34202		43   00000000000000000000000000000000000
81 . X X C 81	662.01	21/	863-650 37771	11111		Partition of the second
15 5 x 5 12	136.61	475	785-113 28060	28060	29	29   consecutive transplatition to the contract to the contrac
12 3 x 5 9	forta)	1	752-901 30173	30173	2/	10011111111111111111111111111111111111
• CX C 5	19401	292	559-280 19679	1,967.9	۴	andenen and and and and and and and and and an
C CXCP		187	379-126 13105	13105		10.
3 x x x	+96.	2	209-261	1925	1	
E CXX3	-11183		064-61-	166.	F.	1111111111111111
•	132.00	-103	142.62-	120	•	***************************************
6. CXC9.	222.	-	000.	9	P _	
66 66 × 8 60	200.	-	000.	•	•	
98888 3 X 3088888	299401		411 6769-129 273994 635	273954	6	

		1		-	-	
FILEAGE BAND	AVERAGE		TOTAL	-	1 4/6	
	- 1		-			
	233.	<sub>6</sub> -	000.	ь-		
51 X X X 15	+26.84	2372	48.934	2372	0	
46 2 X 2 45	230.	5	0000	0	5	
45 ) x > 48	440.819	1876	158.431	1285	6	
42 > X > 39	40.20	2912	162.018	2498	-	•
35 > x > 36	37.580	2830	150.319	2261	-	•
36 2 x 5 33	30.363	763	34.369	109	6	
32 5 x > 30	Brege	861	190.981	96/4	-	
12 XX 23E	201.145	1361	*80.320 ZOR*	20804	-	3 1000
92 CX C 12	164.62	1889	898/2 508-095	898/2		
12 XX X 12	246.34	268	21662 9/6-18/	21662	•	
BI CXCIE	Edb. ci	- 188	698./68	30246	-	
18 3 K 9 . 15	262.01	118	1///5 050-508	311118	-	
15 5 X 5 31	130.361	-	09082 611-584	09082	-	
6 CX C 21	In. di	-	106-264	30173	F	Ti leangageanni li
53 X3 6	16801	292	555.280 19879	13679	-	252222222222222222222222222222222222222
6 5 K 3	299.4	187	379-126 13105	13105	=	13 (2010)
S X X S	+36.	36	132.602	1929	22	A designation of the second of
6. 5863	264.50	64.	064-61-	1600	-	
* C X C E.	122.40	103	142.63-	120	-	
6. CX C 90	- 000		000.	ρ.	•	
66 1 X 3-99999	100		000.	•	P	
99999 3 X 3059999	10001	160	001 +662.13 621.6949 161	1668/2	202	

1	** PILEAGE CVER NOPINAL SLPPARY BY SLPHED PERCENT *** CONFERTS : CHI SAPILES : 1 14 1 24 1 34 44 44 44 44 44 44 44 44 44 44 44 44	CHI SAFLE	FARY BY	1 2, 1 3 187AL	3, V 1,	- N	444 SIPPLE 1 EXD 12 V 22 V 32 F 12 D 24 D 32 N 12 N 22 N 32 A/C I
16 6 6 6 6 6 6 6 6 6 6 6 6 6 6 6 6 6 6		EZ	 	-	5	-	
		_	o -	000.	0 -	0	
			6372		2 -		
					2		
				-	-		
		-		-	-	-	
					53/62		
				160.426	54442	2	
		37.377		710-158	1/262	-	
		330-16)	1432 1	160.478	50136	-	•••
		l -		282-126	17984	-	*****
		-	1193	503.289	107927	-	
	21 2 X 2 18	1	1082	121-146	142129	1	1911121111
		995.22	- 326	844-178	106641	-	***************************************
		222.02	-	169.686	196/02	-	***************************************
	12 3 X 3 9	10.002	-	261.201	20135	06	
	9 (X ( 5	chains	-	305.072	*19/92	-	
	6 2 K 2 3	130mal	-	101-100	216073	-	
	0 (#45	1110061	-	008-618	275167	-	
		ZORIO!	•	0/6-56/		-	
		10.660	164	769-129	13884	991	
10.01		10.000	16.	183.189	13984	001	***************************************
10101	6666-4 X & 6-	10.00	-	109-129	13884	9	
	19995 3 X 3-19999	10101	16.	283-586	1966.	1001	

10C   X   10O				CONTENTS OF THE CAPTURE I I I I I I I I I I I I I I I I I I I			
23.026 1947 46.112 1899	FLELBLEN BAND		.68	2		COUNT	
	9995 X X 2 5000	30.893	9542		30089	1 0	***************************************
	800C 2 X > 1900	\$3.05¢	1947	46.112	3886	~	**
	190C > x > 1800	23.267	1848			0	******
1001   1001	180C > x > 1900	596.92	1948	107-818		-	
23.72C 13.6 13.7376 103.2376 12.73C 1 1 2.73C	1960 3 x > 1600	149-22	1650	67.642	7964	-	•••
25.795   146.4   113.761   782.5   15.275   16.25   15.275   16.25   15.275   16.25   16.275   16.25   16.275   16.2	16CC > x > 1500	29:340	1546	1	10826	-	*******
23.584 12.5 10.5 10.5 10.5 10.5 10.5 10.5 10.5 10	180C > x > 1400		1464	113.761	_	-	******
2011	14CC 3 X > 1300		1961	261-743	19048	=	***************************************
1000  1000	130C 5 x > 1200	285.65	124.9	1	87/8	-	******
19-8C0   1044   217-864   11984   11	18CC 3 X > 1100	822-12	1183		18921	F	***************************************
16001   1600	110C 3 x 5 1000	13.809	1044		18011	=	***************************************
19:123   854   363:48)   16:28   15:28   17:21   16:28   17:21   16:28   17:21   16:28   17:21   16:28   17:21   16:28   17:21   16:28   17:21   16:28   17:21   16:28   17:21   16:28   17:21   17:28   17:	100C 2 x > 800	996.02			*1611	1	
100   10-730   644   459-636   1716.	90C 3 x 5 . 800	181.61	151		10.00	=	
000 1 000 000 000 000 000 000 000 000 0	80C 3 x > 700	289.01	146	1	19141	1	***************************************
	70C 3 x > 600	18.730	*		06981		***************************************
200 1 200 200 200 200 200 200 200 200 20		296.61	266	230.083	09402		
200 1000 240 240 001.100 000 000 000 000 000 000 000 00	80C 5 X > 400	18.91			19135		***************************************
200		add of	348	1	·Cala	1	
100 1 200 200 200 200 200 200 200 200 20	1	982.0	=		18685	=	
96 91916 0	80C \$ X > 100	+92.10	149	1	11808		desenenterenterenterenterenterenterentere
	10C 3 K > 0	elese	*	887.485	61	2	***************************************
erreterinentententententententententent Dr. grafe 1210lage alle 12bere Sallen & C. J. J.	C 2 X 3-33333	1241170			6641	1	***************************************
1	19999 3 X 3-29999	10.00	180	6769-129	1965/2	2	

Date   Principle   Date   Da		CHI SAFFLE	-	1 1 2 1		2	TO : CHI WAPPLES : 1 1. 1 2. 1 3. V 1. V 2. V 3. D 1. C 2. C 3. N 1. N 2. N 3.
1000   20.67.2   27.96   239.598   2009   25.056   19.7   46.112   2895   10.00   25.056   19.7   46.112   2895   10.00   25.056   17.8   10.7818   2988   10.00   25.056   17.8   10.7818   10.8	PLELBLAN BAND	AVERAGE	- 887	NH TOTAL	-	3/4	
100   24.050   1845   16.036   2288   1900   22.557   1845   167.818   6982   1900   22.557   1845   167.818   6982   1900   22.557   1846   113.761   7323   1846   113.761   7323   1846   113.761   7323   1846   113.761   7323   1846   113.761   7323   1846   113.761   7323   1846   113.761   7323   1846   113.761   7323   1846   113.761   1	0008 4 X 4 50		27.56	339.598	30099	~	•
100   24.257   1845   116.036   9228   1700   26.353   1748   107.818   6922   1700   26.354   167.818   6922   1700   26.755   1465   167.818   1700   26.755   1465   167.819   1700   26.755   1465   167.819   1700   26.755   1163   293.812   1200   26.755   1163   293.812   1200   26.755   1163   293.812   1200   26.755   1163   293.812   1200   26.755   1200   26.755   1200   26.755   1200   26.755   1200   26.755   1200   26.755   1200   26.755   1200   26.755   1200   26.755   1200   26.755   1200   26.755   26.7	-	23.086	1947	46.112	3696	0	
100   28.587   1948   107.818   6982     100   28.587   1656   67.642   4982     100   28.782   1844   113.761   7883     100   28.782   1844   113.761   7883     100   28.782   1844   113.761   7883     100   28.782   1844   113.761   7883     100   28.782   1843   167.814   1788     100   28.782   1843   167.814   1788     100   28.782   1843   167.814   1788     100   28.782   1843   1848     100   28.782   1844   1848     100   28.782   1844   1848     100   28.782   1844   1848     100   18.782   644   48.812   1848     100   18.782   644   1848     100   18.782   1848     100   18.782   1848     100   18.782   1848     100   18.782   1848     100   18.782   1848     100   18.782   1848     100   18.782   1848     100   18.782   1848     100   18.782   1848     100   18.782   1848     100   18.782   1848     100   18.782   1848     100   18.782   1848     100   18.782     100   18.782     100   18.782     100   18.782     100   18.78		23.267	1848	116.036	1226	-	•
100   24.54   165   67.642   498   150   24.75   166   67.642   498   166   67.642   498   166   67.642   498   166   67.642   498   166   67.642   498   166   67.642   498   166   67.642   498   166   67.642   498   166   67.642   498   166   67.642   498   166   67.642   498   166   67.642   498   166   67.642	3C > x > 1700	596.03	1.4.	107-818	2669	-	
100   24-752   1464   113-76   7782   100   24-752   1464   113-76   7782   100   24-752   1464   113-76   7782   100   24-752   1464   113-76   7782   100   24-752   1164   24-752   1164   1164   24-752   1164   24-752   1164   24-752   1164   24-752   1164   24-752   1164   24-752   1164   24-752   1164   24-752   24-752   1164   24-752	DC 3 x > 1600	28.547	1650	67.642	4952	0	
100   24.752   1464   113.761   7323   15008   1500   24.750   1368   261.743   15008   1500   27.252   1153   250.552   250.552	CC 3 x > 1500	23.346	1906	163.378	10886	-	
100	00 5 X > 1400		1464	113.761	1383	-	
100		23.795	1361	261.743	19048	~	
100	13CC > X > 12CO	23.588	124.9	167.914		-	
1000   19:800   1044   217:804   11984   1000   19:125   105   1	186C > x > 1100	-	1163	219.863	18021	~	
100   20.366   948   281.391   11314   1523   1534   1334   1334   1334   1334   1334   1334   1334   1334   1334   1334   1334   1334   1334   1334   1334   1334   1334   1334   133	110C > x > 10C0	13.800	1000	217-854	18011	~	
100   10.129   194   363.4597   16229   1761   18090	100C 2 x > 300	50.986	246	251.591	P. IEII	~	=
100   10.68C   746   426.35   7161   600   10.735   644   456.35   1890   600   10.735   644   456.35   1890   600   10.735   645.35   1890   600   10.736   456.35   12.34   600   10.736   456.35   12.34   600   6.736   6.735	008 . 4 × 4 326	19-125	16	363.457	16229	-	
000 10.750 644 486.312 18990 090 090 090 090 090 090 090 090 090		18.680	146	429-636	17161	•	****
100   12.867   468   542.976   19186		18.735	**	456.312	18690	6	
000 10-200 200 000000 10130 000 000 000 000 000 000		296.61	298	530.583	09402	•	
300 10-246 346 637-703 21434   150-103   150-1		189.25		948-346	19135	-	1999111
100		10.286	200	637-763	16412	10	***************************************
100 8-264 145 636-364 11208 0 3-510 36 597-496 6133 099 61-621 639 501-171 61-515 1999 10-660 431 6769-1289 276964		962.0	1	412-815		•	***************************************
0 3-510 36 597-495 6133 999 -1-421 -39 -01-171 -1435 1999 10-561 431 6769-189 276984	1	192.0	140	+91.919	11208		***************************************
999 10:000 431 0703:127 274350 1	10C 3 x 5 0	2.53.2	*	587.495		4	nangananananan da manananan da manananan da mananan da
299.01 666	C 5 X 5-59999	130.50	6.	1/1-19-		•	***************************************
	95 2 X 3-39999	10.01	16.	621-69/9	196812	200	

	AVERAGE	- 59,	TOTAL	=	*	
3 X > 8000	30.893	2736	339.598 30099	30099	-	•
3 x > 1900	29.67	8615	365-710 33995	33995	~	•
3 x > 1800	27.075	20c1	501.746 +322+	+385+	•	3 ···
C > x > 1700	136.18	2382	609-564 50216	90216	-	3 45
C > x > 1600	87.086	9022	677-206 55169	69169	•	•••
C 3 x > 1500	50.268	2902	840-584 69995	56669	5	••••
C > X > 1400	28.793	1961	954.345 73319	13319	•	****
C > x > 1300	26.335	1840	1216.088	88367	•	*****
C > x > 1200	PD-164	1963	1765 1384.002 97116	97116	•	9   00000
C > x > 1100	236.02	16.63	108801 +1683-914 103804	109801	2	10   ++++
C > x > 1000	56.05	1575	1901-368 121289	121289	12	12   000000
006 K X C D	161-42	1489	1489 2152-959 132604 14   seeses	132604	=	*******
C 3 x 5 . 800	236.02	1375	1978 FB16.416 148834	14.00.1	=	
C 3 x > 700		1867	5646-052 165995	166999	12	12
009 < x c 3	292.12	1184	1154 3402-364 184686	184686	1	92
C > X > 800	\$92.08	1641	1041 3992-947 205146	941502	-	31
00 × × × 00	19.80	*6	4535-923 224285	284422	=	
098 ×× 5	161-/1	2	813 5173.626 249716	91/642		***************************************
098 < X C D	160.01	2	20062 144-9856	210862	6	90.
C > x > 100	10.60	689	6222.805 269256	992692	=	
10C > X > 0	13.380	684	489 68E0-300 275389	275389	*	
C 2 x 2-99999	10.01	164	6769-129	198672	6	+91 6769.129 273964 100   10: 10: 10: 10: 10: 10: 10: 10: 10: 10:
19 3 X 3-89999	10.00	16.	481 6769-129 273984 100	1988/2	2	
	-					

NA ERAGE LES 1 1701AL PA-CEZ 3968 1110-336 E	•	
96-98- 2910 117-364	LBS   COUNT!	
\$6.98¢ 291C 117.964	53479 15 100	12
	5820 2 100	
44.610 2739 97.232	5479 2 00	
85-545 2652 13-549	2692	
87-833 E44C 173-498	7321 3 1000	
.S. Ect 1323 299.164	13942 6	****
2 x > 2100 38-368 2175 387-575 19230	5230 7 0026	
2 x y 1950 36-101 2022 190-503 10114	C114 5   *****	
35-428 1848 319.391	16819 9	6
3 x y 1650 40-159 1708 369-273 13665	3665 . 8	******
42.826 1567 642.383	23519 15   000	15
2 K > 1350 30.405 1414 459.133 21213		15
3 x 5 .1200 34.526 1273 1141.322 42039		23   Section Contraction Contr
180c > x > 1050 3x-795 1135 688-690 23850	-1 -	72
2 x > 900 20.532 979 587.230 22533	1	23   100000000000000000000000000000000000
90C > x > 750 E4.832 835 1018.108 34257		it ferrengeneuten frementation and an annual statement
3 x 3 600 17-595 670 737-303 28153		the transmitter to the transmitt
092/2 29/-16/ 184 25 25001 084 CX C		25 January and Company of the Compan
28602 909-824 091 18:880 380 XX 300 XX		Redicate de la constante de la
N N 150 10-312 215 680-592 14243		224520222202222222222222222222222222222
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### N. P.	COMENTS :	FUEL BUMPARY		I IS I SO I BO V IS V SO V BO	3. V 1. V 2	E v 3s n 1s D E v D 3s h 1s h Es h 3s	
1,364   256   1110-334   59479   1416   252   14549   14522   14579   14549	FLELDLAN BAND	AVERAGE		TOTAL	-		
1362   2510   117-364   5820   1515   252   25	99995 X X 3000	74.022	1965	1110.336	59479		
154   275   57.232   64.99   155.4	300C 3 X > 2850	28.36	2910	117.964	5820	10	
185.1   262   18.55.3   2692   18.55.3   2692   18.55.3   2692   18.55.3   2692   18.55.3   2692   18.55.3   2692   18.55.3   2692	885C > X > 2700	*****	2739	97.232	8479		
183	870C 2 x > 2550	69:243	1692	13.549	2692	0	
126   232   295.164   13912   13613	866C 2 X > 8400	87.833	2440	173.498	1364	0	
126	840C 3 X > 8250	198-60	1323		13842	•	
1416. 2622 190.503 10114 1418 1848 319.391 16819 1418 1703 389.273 13889 1418 1203 1811322 42039 1418 1213 680.593 22833 1418 1219 680.593 22833 1418 1219 680.593 22833 1418 1219 680.593 14283 1418 1219 141832 14283 1418 1219 141833 14283 1418 1219 141833 14183 1418 1219 141833 14183 1418 1219 141833 141833 1418 1219 141833 141833 1418 1219 141833 141833 1418 1219 141833 141833 1418 1219 141833 141833 1418 1219 141833 141833 1418 1219 1219 12183	225C > x > 2100	296	2175	347.575	19230	1 0	
186   186   189	81CC > X > 1950	30-101	2002	190.503	16114	• •	
126	1960 2 X V 1800	-	1981		16919		
1.65 154 485.133 21213 1.65 1414 485.133 21213 1.68 1273 1141.322 42039 1.53 113 686.590 28880 1.53 113 686.590 28880 1.53 113 686.590 28880 1.53 113 686.590 28880 1.53 113 686.590 28883 1.53 113 686.590 28883 1.53 113 686.590 28883 1.53 113 686.590 28883 1.53 113 686.590 28883 1.53 113 686.590 28883 1.53 113 686.590 28883 1.53 113 686.590 28883 1.53 113 686.590 28883 1.53 113 113 113 113 113 113 113 113 113 1	1800 > x > 1650	681.	1761	369.273	13665	•	
100   1014   680-133   21213   11213	165C > x > 1800	939-24	1567	648.383	STARE	*   ••	
135   1141-322   42039   135   136   42030   23800   135	196C > x > 1350	30.00	1010		21213	•••	
1532 979 987.20 22893 1532 979 987.20 22893 1532 879 1016.108 34287 15342 824 797.762 22893 15342 824 797.762 27897 15342 824 797.762 27897 15342 819 1489.789 14883 1532 819 1489.899 14883 1532 819 1489.899 14883 1543 811 876.999 14883 1543 81 876.999 14883 1543 81 876.999 14883	138C 5 x 5 .1200	989-46	1893	1101.322	42039	999901 9	
1932 999 587.230 22933 1932 839 1016.108 34287 1950 670 737.303 28183 1950 870 727.303 28183 1950 870 727.808 2028 1950 81 876.989 1489 1952 81 876.989 1489 1952 81 876.989 1489 1953 81 876.989 1489 1953 81 876.989 1489	180C 3 X > 1050	38.798	1138	069-889	23880	••••	
1.536 670 737-303 28193 1-346 670 737-303 28193 1-346 584 737-786 27280 1-312 213 680-592 14843 1-312 213 680-592 14843 1-312 213 680-592 14843 1-313 613 1487-334 41840	105C > x > 900	269.63	979		22833	*	
1950 670 737-303 E8183 1950 884 737-78 2780 1950 860-898 14843 1932 81 876-898 14843 1932 81 876-898 14843 1933 91 819-834 41840	860 3 X > 780	268-12	138	1018-108	34287		
10   10   10   10   10   10   10   10	78C 3 X > 600	17.555	690	137.303	C8182	***************************************	
1-312 E15 640.031 1483 1-312 E15 640.031	60C 2 X 2 450	20Coal	524	197.767	27260	********	
1001	48C 8 X 9 300	18096	260	728-806	2022	announce &	
1001	30C 3 X > 150	216.01	6113	269-089	14843	************ Di	
1021 - 620 11487.137 - 1301	18C 3 x 3 0	/26	16	176.969	3606	***************************************	
1.043 6g5 [[487-336 4]8308	C 2 X 3-99999	elly 10			1981	*******	
	99999 5 X 5-99999	10.01	989	11487.336	-1	1 90	

		10   1   10   10   10   10   10   10		Fighting and   Average   11, 12, 12, 13, 14, 17, 17, 17, 17, 17, 17, 17, 17, 17, 17	0 82 D 32 h 32 h 32 h 32																•	***************************************	***************************************	***************************************	Transportation transportation to the state of the state o	67 innentimentationalparentations	Transcrappentrerererennennententententententententententente	estinational and property of the contract of t		•
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120CC   X   120CO	COMPENTS :	4	1 : 3	1 8 1	3. 4 1.	2 >	Er 1 30 V 10 V 20 V 30 F 10 D 20 D 30 F 10 H 20 F 30
1000   0000   0   0   0   0   0   0	ALTITUDE BAND	AVERAGE	1.85	TOTAL	=	A.C.	
1134   10.823   13.8	9995 2 x > 12800	300.	-0	900.	-	0	
1-194	880C 2 X > 18000	3:438	=	3.438	=	•	•••••
11194	860C 3 X > 11500	908-6	66	19-823	52	•	*******
0.00   0.000	166C > x > 11000	1.194	:	3.881	=	~	•
7:174   239   96.910   9069     7:174   215   46.564   87.8     7:174   215   46.564   87.8     7:174   215   46.564   87.8     7:175   26.95   42.95   42.95     7:175   26.95   42.95   42.95     7:175   26.95   42.95   42.95     7:175   26.95   42.95   42.95     7:175   26.95   42.95   42.95     7:175   26.95   42.95   42.95     7:175   26.95   42.95   42.95     7:175   26.95   42.95     7:175   26.95   42.95     7:175   26.95   42.95     7:175   26.95   42.95     7:175   26.95   42.95     7:175   26.95   42.95     7:175   26.95   42.95     7:175   26.95   42.95     7:175   26.95   42.95     7:175   26.95   42.95     7:175   26.95   42.95     7:175   26.95   42.95     7:175   26.95	100C 3 X > 20800	•36•	120	13.687	092	=	***************************************
7.174   275   64.566   8475     3.537   107   38.907   1178     4.847   226   32.032   4294     4.847   226   226.234   1078     4.847   236   226.234   1078     4.847   236   226.234   1078     4.847   236   226.234   1078     4.847   236   1123.735   4190     4.847   236   1123.735   4190     4.848   23.035   4190     4.848	000C 3 x > 10000	169.6	833	014.96	1065	10	***************************************
00	000C 1 X > 9800	10194	878	84.564	2473	=	***************************************
00	980C 2 x > 9000	3.63)	107	38.907	1178	-	*******
00   0.	9000 X X 9 8800	(98.	928	35.092	1621	-	2000000
00   9-54C   598   485.534   14197   00   0-3C1   291   771-370   27091   00   1-4C7   593   1123-759   41020   00   1-4C7   590   1123-759   41020   00   1-4C7   590   1123-759   4200   00   1-4C7   590   1233-759   4200   00   1-4C7   593   237-879   2241   00   1-4C7   593   237-879   2241   00   1-4C7   593   237-879   2241   00   0-377   264   1-4C9   237-879   2420   00   0-377   264   247-879	100 × × 2000	1.763	962	163-543	10732	4.5	***************************************
00   0.3C1   291   771.970   27031   00   0.2C4   373   788.3E6   31030   00   1.0C4   312.759   31030   00   1.0C4   312.759   32021   00   1.0C4   312.759   32021   00   1.0C4   312.759   32021   00   1.0C4   312.759   32021   00   0.0C4   312.759   32021   00   0.0C4   312.759   32021   00   0.0C4   312.759   32021   00   0.0C4   312.759   00   0.0C4   312.759   00   0.0C4	60C 2 x > 7500	3.540	250	457.936	16191	1.8	***************************************
	200 X X 7000	136.4	168		16043	1.5	***************************************
100   1**0;   530   1123-759   1102   100   10	780E > x > 6930	182.4	2	326-896	21030	=	***************************************
100   1*123   746   8257035   88601   100   1*122   746   8257035   88601   100   1*122   746   437799   32481   100   1*122   746   437799   32481   100   4*70   32481   100   4*70   32481   100   4*70   100	PECE > x > 6000	10.467	930	1183-759	20114		Paranta Parant
10   10   10   10   10   10   10   10	100C 3 X 5 8800	14.034	240	1263.035	10984	92	***************************************
100   14:129   744   437-999   20101   100   1	0008 X X 2 2000	16.01	1	1820737	MOE	2	***************************************
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	000 4 X 2 000	18.88		217.878	1		neutroniteteren er
	+60C 5 x > 3800	63/00	-	108-19	1	=	- Constitutions
99 - 10-10-1   1-10   10-10-1   10-1	980C 3 x 5 3000	148.0	282	121-159	1	1	***************************************
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1180C 3 x > 11000 1-194	••	3.581	261	2		
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\$80C 2 X > 9000 3-637	101 (65	18.907	1178	•	*******	
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NON LEADING THE NOMINAL PAIR CATA FILE!  CAI KYNOW RADM 11:03 NEV 14/177 12 BAMPLES FUELFLOW DATE 22:06 MAR 14/178  CAI KYNOW RADM 11:03 NEV 16/177 12 BAMPLES FUELFLOW DATE 22:06 MAR 14/178  THE FELLENING BAMPLES MANE BEEN SELECTEC FOR SUPPARIZATION!								

ACCULATION FOR MANY CONTRACTOR OF THE CONTRACTOR	VERBION CO3
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